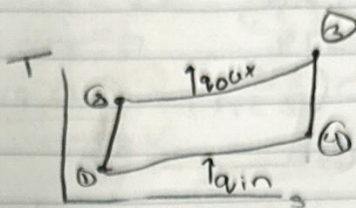
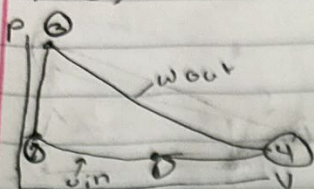


Aaligah Ashley

HW 1.2

9-13)



$$273 + 22 = 295 \text{ K} \quad C_p = 1.005 \text{ kJ/kg} \cdot \text{K} \quad C_v = 0.718 \text{ kJ/kg} \cdot \text{K}$$

$$R = C_p - C_v = 0.287 \text{ kJ/kg} \cdot \text{K} \quad \gamma = C_p / C_v = 1.4$$

State 1 Isentropic Compression State 2 V constant State 3 Isentropic State 4 P constant

$$P_1 = 100 \text{ kPa}$$

$$P_2 = 600 \text{ kPa}$$

$$T_3 = 1500$$

$$P_4 = 100 \text{ kPa}$$

$$T_1 = 295 \text{ K}$$

$$v_1 = \frac{RT_1}{P_1} = \frac{(0.287 \text{ kJ/kg} \cdot \text{K})(295)}{100} = 0.847 \text{ m}^3/\text{kg}$$

$$\frac{v_2}{v_1} = \left(\frac{P_1}{P_2} \right)^{\frac{1}{\gamma}} = \left(\frac{100}{600} \right)^{\frac{1}{1.4}} = 0.1412 \text{ m}^3/\text{kg}$$

$$\frac{P_3}{P_2} = \frac{T_3}{T_2}$$

$$\frac{v_4}{v_3} = \left(\frac{P_3}{P_4} \right)^{\frac{1}{\gamma}} = \left(\frac{100}{1500} \right)^{\frac{1}{1.4}} = 0.276$$

$$h_1 = 295.17 \text{ kJ/kg}$$

$$\frac{T_2}{T_1} = \left(\frac{v_1}{v_2} \right)^{\gamma} = \left(\frac{0.847}{0.1412} \right)^{1.4} = 604 \text{ K}$$

$$P_3 = \frac{1500}{600} (600) = 1510 \text{ kPa}$$

$$\frac{T_4}{T_3} = \left(\frac{v_3}{v_4} \right)^{\gamma} = \left(\frac{0.276}{0.0182} \right)^{1.4} = 1500 \text{ K}$$

$$u_2 = 438.1$$

$$v_3 = \frac{(0.287)(1500)}{1510} = 0.276$$

$$T_4 = \frac{(0.276)^{0.4}}{(0.0182)^{0.4}} (1500) = 1500 \text{ K}$$

$$u_3 = 1205.41 \text{ kJ/kg}$$

$$b) W_{net} = W_{out} - W_{in}$$

$$W_{out} = \frac{0.287 (4443.85 - 1500)}{1 - 1.4} = 2112.21$$

$$W_{in} = \frac{0.287 (604 - 295)}{0.4} = 221.71$$

$$W_{net} = 1890.5$$

$$c) 0.718 \text{ kJ/kg} (1500 - 600) = 646.2$$

$$\eta_{th} = 0.341 \text{ or } 34\%$$

9-18) $T_1 = 350 \text{ K}$ $T_4 = 1200 \text{ K}$
 $P_1 = 150 \text{ kPa}$ $P_4 = 300 \text{ kPa}$
 $P_{r1} = 2.379$ $P_{r4} = 238$
 $300 \left(\frac{238}{2.379} \right) = 30 \text{ MPa}$

$W_{out} = 0.5 \text{ kJ}$

$\eta_{th} = 1 - \frac{Q_L}{Q_H} = 1 - \left(\frac{350}{1200} \right) = 0.708$

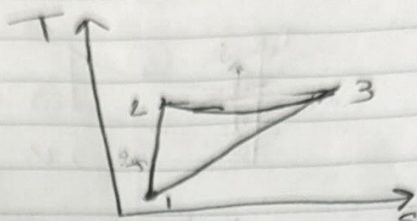
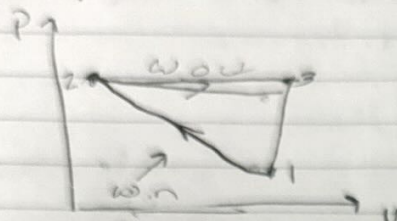
$\frac{W_{out}}{\eta_{th}} = \frac{0.5}{0.708} = 0.706 \text{ kJ}$

$0.287 \ln \left(\frac{300}{50} \right) = 0.1989$

$(1200 - 350) (0.1989) = 169.093 \frac{\text{kJ}}{\text{kg}}$

$\frac{0.5}{169.093} = 0.00295 \text{ kg}$

9-12)



$$\gamma = 5 \quad C_v = 0.7 \text{ kJ/kg} \cdot \text{K}$$

$$R = 0.3 \text{ kJ/kg} \cdot \text{K}$$

$$C_p = 1 \quad \gamma = 1.43$$

State (1)

State (2)

State (3)

$$T_1 = 293 \text{ K}$$

$$\frac{T_2}{T_1} = (5)^{1.43-1}$$

$$\frac{T_3}{T_1} = (5)^{1.43} = 2926.8 \text{ K}$$

$$T_2 = 585.36 \text{ K}$$

$$w_{1-2} = \frac{0.3(585.36 - 293)}{1 - 1.43} = -203.97 \text{ kJ/kg}$$

$$Q_{2-3} = C_p(T_3 - T_2) = 1(2926.8 - 585.36) = 2341.44 \text{ kJ/kg}$$

$$w_{2-3} = 0.3(2926.8 - 585.36) = 702.43 \text{ kJ/kg}$$

$$q = 0.7(2926.8 - 293) = 1843.66$$

$$1 - \left(\frac{1843.66}{2341.44} \right) = 0.2125 = \boxed{21.39\%}$$

$$\eta = 1 - \frac{(T_3 - T_1)}{K(T_3 - T_2)}$$

$$= 1 - \frac{\left(1 - \frac{T_1}{T_3}\right)}{K \left(1 - \frac{T_2}{T_3}\right)}$$

$$= \frac{1 - (r^{\frac{1}{\gamma}} - 1)}{K(r - 1)}$$

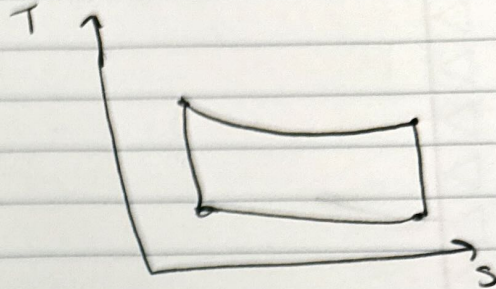
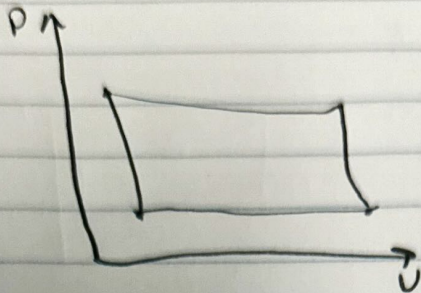
9-31)

$$r = 10.5$$

$$P = 90 \text{ kPa}$$

$$T = 313 \text{ K}$$

$$\text{Power} = 90 \text{ kW}$$



$$1 - (10.5)^{-0.4} = 0.6095$$
$$\boxed{60.95\%}$$

$$W = 90 \times \frac{60}{2500}$$

$$= 2.16 \text{ kJ}$$

$$\frac{2.16}{0.6095} = \boxed{3.543 \text{ kJ}}$$