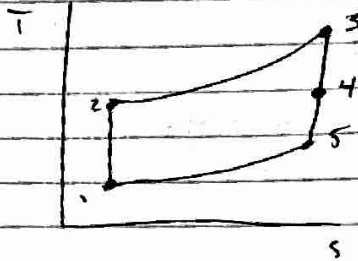


135



① A-17 7°C 280°K
 $h_1 = 280.13 \text{ kJ/kg}$
 $P_{r1} = 1.0889$

$$\frac{P_2}{P_1} = \frac{P_{r2}}{P_{r1}} \quad P_{r2} = P_{r1} \left(\frac{P_2}{P_1} \right) \quad P_{r2} = 1.0889 (9)$$

$$P_{r2} = 9.8001$$

Enthalpy @ P_{r2} A-17

$$h_2 = \frac{(9.8001 - 9.684)(533.48 - 523.63)}{(10.37 - 9.684)} + 523.63$$

$$= 525.38 \text{ kJ/kg}$$

$$\dot{Q}_{in} = .5(427000)$$

$$= 21350 \text{ kW}$$

$$\dot{q}_{in} = \frac{\dot{Q}_{in}}{\dot{m}_a} = \frac{21350}{20} = 1067.5 \text{ kJ/kg}$$

$$q_{in, 2-3} = h_3 - h_2 = 1067.5 = h_3 - 525.38$$

$$h_3 = 1592.8$$

T_3 Relative pressure @ h_3

$$T_3 = \frac{(1592.8 - 1587.63)(1480 - 1460)}{1611.79 - 1587.63} + 1460$$
$$= 1464.34^\circ\text{K}$$

$$Pr_3 = \frac{(1592.8 - 1587.63)(568.8 - 537.1)}{1611.79 - 1587.63} + 537.1$$
$$= 543.9$$

W by turbo

$$W_t = W_c$$
$$h_3 - h_4 = h_2 - h_1$$
$$1592.8 - h_4 = 525.38 - 280.13$$
$$h_4 = 1347.63 \text{ kJ/kg}$$

$$\frac{Pr_3}{Pr_5} = \frac{P_3}{P_5} \quad \frac{543.9}{Pr_5} = 9 \quad Pr_5 = 60.4$$

Find Enthalpy A-17 @ Pr_5

$$h_5 = 877.51 \text{ kJ/kg}$$

Apply SFEE Equation

$$h_4 + \frac{1}{2}V_i^2 = h_5 + \frac{1}{2}V_E^2$$

$$V_E^2 = \sqrt{2(h_4 - h_5)} = 969.59 \text{ m/s}$$

$$F = \dot{m}_a(V_E - V_i)$$
$$20(969.59)$$

$$= 19391.95 \text{ N}$$

