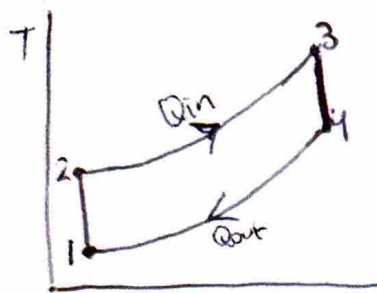
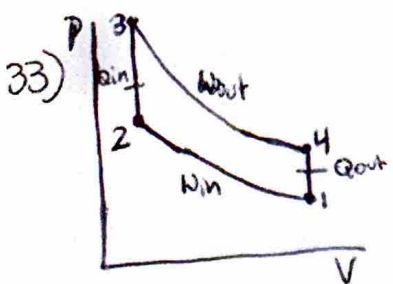


Ramzie Abbas

HW 1.3 - 33, 36, 46, 57, 59, 80



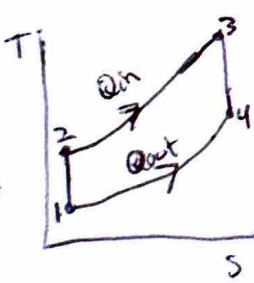
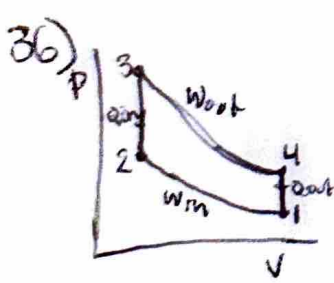
$T_1 = 300K$
 $P_1 = 95 kPa$
 From A-17
 $u_1 = 214.07 \frac{kJ}{kg}$
 $v_{r1} = 621.2$
 $v_{r2} = \frac{v_{r1}}{cr} \text{ comp ratio}$
 $= \frac{621.2}{8}$
 $v_{r2} = 77.65$
 $T_2 = 673K$
 $P_2 = ?$
 $P_2 = P_1 \cdot \frac{T_2}{T_1}$
 $= 95 \cdot \frac{673}{300}$
 $P_2 = 1705 kPa$
 $u_2 = 491.2 \frac{kJ}{kg}$
 $v_{r2} = 77.65$
 $T_3 = 1539K$
 $P_3 = 3899 kPa$
 $v_{r3} = 6.588$
 $\Delta u_{2-3} = q_{in}$
 $u_3 = u_2 + q_{in}$
 $u_3 = 1241.2 \frac{kJ}{kg}$
 From A-17 we can get T_3
 $T_3 = 1540K$
 $v_{r3} = 6.588$
 $P_3 = P_2 \cdot \frac{T_3}{T_2}$
 $= 1705 \cdot \frac{1540}{673}$
 $P_3 = 3899 kPa$
 $T_4 = 780K$
 $u_4 = 576.12 \frac{kJ}{kg}$
 $v_{r4} = 52.7$
 $v_{r4} = cr \cdot v_{r3}$
 $= 8 \cdot 6.588$
 $= 52.7$
 from A-17 we can get $T_4 = 780K$
 $u_4 = 576.12 \frac{kJ}{kg}$

a) $T_3 = 1539K$
 $P_3 = 3899 kPa$

b) $w = q_{in} - q_{out}$
 $= q_{in} - (u_4 - u_1)$
 $= (750 - (576.12 - 214.07))$
 $w = 387.95 \frac{kJ}{kg}$

c) $\eta = \frac{w}{q_{in}} = \frac{387.95}{750}$
 $\eta = 0.517 \text{ or } 52\%$

d) $MEP = \frac{8 \cdot 387.95 - 95}{0.287 \cdot 300(8-1)}$
 $MEP = 495 kPa$



isentropic $\rightarrow$ $T_1 = 565R$
 $P_1 = 14 psia$
 $T_2 = 1430.5R$
 $T_3 = 2860R$
 $T_4 = 1129.6R$

$T_2 = T_1 r^{k-1}$
 $= 565 \cdot \left(\frac{1}{0.008}\right)^{1.4-1}$
 $= 1430.5R$

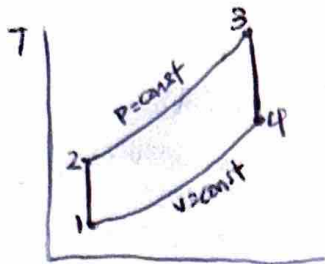
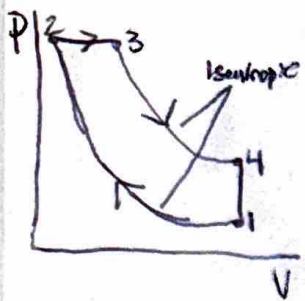
$T_4 = T_3 r^{1-k}$
 $= 2860 \cdot 0.008^{1-1.4}$
 $= 1129.6R$

$M = \frac{P_1 N \pi B^2 S}{4RT_1}$
 $= \frac{14 \cdot 6 \cdot \pi \cdot 3.5^2 \cdot 39 \cdot \left(\frac{1}{12}\right)^3}{4 \cdot 0.3704 \cdot 565}$
 $= 8.72 \cdot 10^{-3} \text{ lbm}$

$W = \frac{mncv}{N_{rev}} (T_3 - T_4 - (T_2 - T_1))$
 $= \frac{8.72 \cdot 10^{-3} \cdot 2500 \cdot 0.171}{60 \cdot 2 \cdot 0.7068} (2860 - 1129.6 - (1430.5 - 565)) \text{ hp}$

$W = 38 \text{ hp}$

46)



$$\eta = 1 - \frac{q_{out}}{q_{in}}$$

$$= 1 - \frac{u_4 - u_1}{h_3 - h_2}$$

$$= 1 - \frac{659.69 - 214.07}{1910.6 - 890.89}$$

$$\eta = 0.563 \text{ or } 56.3\%$$

$$MEP = \frac{0.563(1910.6 - 890.89)}{0.287 \cdot 300 \left(1 - \frac{1}{16}\right)}$$

$$MEP = 675.7 \text{ kPa}$$

Isentropic $p = \text{const}$ Isentropic $v = \text{const}$

1 2 3 4

$T_1 = 300.15 \text{ K}$ $T_2 = 862.4 \text{ K}$ $T_3 = 1724.8 \text{ K}$ $T_4 =$

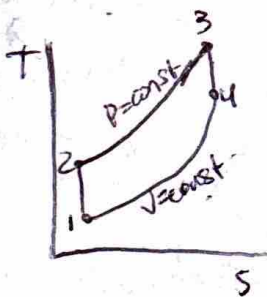
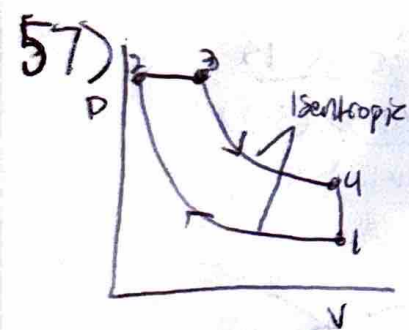
$P_1 = 95 \text{ kPa}$ $h_2 = 890.89 \text{ (from A-17)}$ $h_3 = 1910.6 \text{ KJ/kg}$ $v_{m1} = 36.37$

$u_1 = 214.07 \text{ KJ/kg}$ $v_{r2} = 38.825$ $v_{r3} = 4.546$ $u_4 = 659.69$ $v_{r4} = \frac{v_3}{r_c}$

$v_{r1} = 621.2$ $v_{r2} = 38.825$ $T_3 = T_2 \cdot \frac{v_{r2}}{v_{r1}} = 862.4 \cdot \frac{38.825}{621.2} = 530.4 \text{ K}$

$T_3 = 1724.8 \text{ K}$

$v_{r4} = \frac{16}{2} \cdot 4.546 = 36.37$



Isen $p = \text{const}$ Isen $v = \text{const}$

1 2 3 4

$T_1 = 343.15 \text{ K}$

$P_1 = 97 \text{ kPa}$

$$W = \eta \cdot q_{in}$$

$$= \left(1 - \frac{1}{k} \frac{r_c^k - 1}{r_c^{k-1}(c-1)}\right) \phi T_1 r^{k-1} (c-1)$$

$$= \left(1 - \frac{1}{1.4} \frac{1.8^{1.4} - 1}{22^{1.4-1} (1.8-1)}\right) \cdot 1.005 \cdot 343 \cdot 22^{1.4-1} (1.8-1) \frac{\text{KJ}}{\text{kg}}$$

$$= 635 \frac{\text{KJ}}{\text{kg}}$$

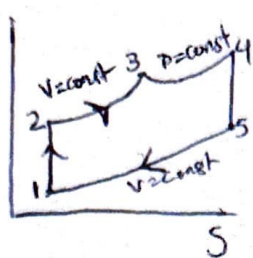
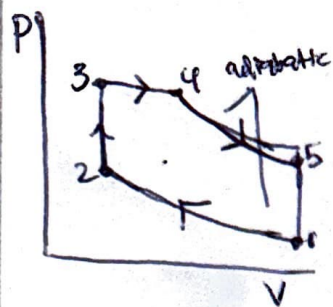
$$W = \dot{m} w$$

$$= N \frac{P_1 V}{RT_1} w$$

$$= \frac{3500}{60} \frac{97.25 \cdot 10^{-3}}{0.287 \cdot 343} \cdot 635 \text{ kW}$$

$$= 91.2 \text{ kW}$$

59)



① ② ③ ④ ⑤

$T_1 = 535R$
 $P_1 = 14.2 \text{ psia}$

$$q_{in} = c_v T_1 r^{K-1} \left(\frac{P_1}{P_2} - 1 \right) + c_p \frac{P_1}{P_2} T_1 r^{K-1} (r_c - 1)$$

$$= (0.171 \cdot 535 \cdot 15^{1.4-1} (1.1 - 1) + 0.29 \cdot 1.1 \cdot 535 \cdot 15^{1.4-1} (1.4 - 1)) \frac{\text{Btu}}{\text{lbm}}$$

$$q_{in} = 193.9 \frac{\text{Btu}}{\text{lbm}}$$

$$\eta_{th} = \frac{w}{q_{in}} = \frac{124.2}{193.9}$$

$$\eta_{th} = 0.646 \text{ or } 64.6\%$$

$$w = q_{in} - q_{out}$$

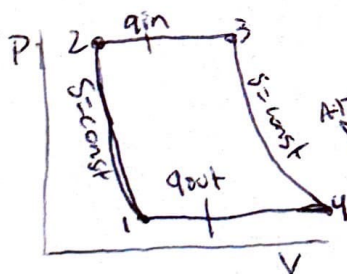
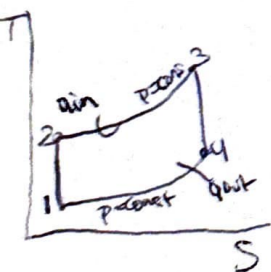
$$= q_{in} - c_v (T_4 - T_1)$$

$$= q_{in} - c_v T_1 \left(r_c^K \frac{P_1}{P_2} - 1 \right)$$

$$= (193.9 - 0.171 \cdot 535 (1.4^{1.4} \cdot 1.1 - 1)) \frac{\text{Btu}}{\text{lbm}}$$

$$w = 124.2 \frac{\text{Btu}}{\text{lbm}}$$

80)



① $T_1 = 520R$
 $P_1 = 1.2147$
 $h_1 = 124.27 \frac{\text{Btu}}{\text{lbm}}$

② $T_2 = 996R$
 $P_2 = 12.147$
 $h_2 = 240.11 \frac{\text{Btu}}{\text{lbm}}$

③ $T_3 = 2000R$
 $P_3 = 17.4$
 $h_3 = 504.71 \frac{\text{Btu}}{\text{lbm}}$

④ $P_4 = 17.4$
 $h_4 = 265.83 \frac{\text{Btu}}{\text{lbm}}$

$$P_{r2} = r_p P_1$$

$$= 10 \cdot 1.2147$$

$$P_{r2} = 12.147$$

$$P_{r4} = \frac{P_{r3}}{r_p}$$

$$= \frac{17.4}{10}$$

$$= 1.74$$

Backwork Ratio:

$$r_{bw} = \frac{w_{in}}{w_{out}}$$

$$= \frac{h_2 - h_1}{h_3 - h_4}$$

$$= \frac{240.11 - 124.27}{504.71 - 265.83}$$

$$r_{bw} = 0.485$$

$$\eta = \frac{w}{q_{in}} = \frac{w}{h_3 - h_2}$$

$$= 1 + \frac{h_1 - h_4}{h_3 - h_2}$$

$$= 1 + \frac{124.27 - 265.83}{504.71 - 240.11}$$

$$= 0.465$$

$$\eta_{th} = 0.465$$