



a) $P_4 = 300$

$$P_1 = P_4 \frac{P_{r1}}{P_{r4}}$$

$$= 300 \cdot \frac{2.38}{2.379} \text{ kPa}$$

$$= 300.13 \text{ kPa} = \boxed{30 \text{ MPa}}$$

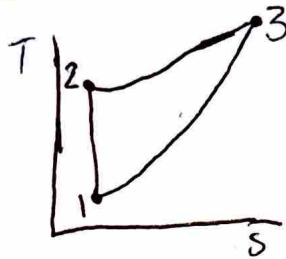
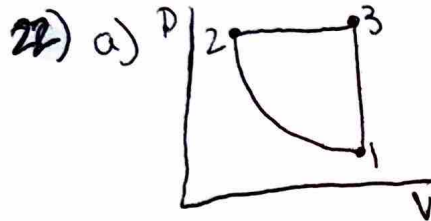
b) $\eta = 1 - \frac{T_L}{T_H}$

$$\frac{W_{net}}{Q_{in}} = 1 - \frac{T_L}{T_H}$$

$$Q_{in} = \frac{T_H W_{net}}{T_H - T_L}$$

$$= \frac{1200 \cdot 0.5}{1200 - 300} \text{ kJ}$$

$$= \boxed{0.706 \text{ kJ}}$$



$$T_1 = 293$$

b) 2-3

$$q_{2-3} = W_{2-3} + \Delta U_{2-3}$$

$$= (R + C_v) T_1 r^{R/C_v} (r - 1)$$

$$= (0.3 + 0.7) \cdot 293 \cdot 5^{0.3/0.7} (5 - 1) \frac{\text{kJ}}{\text{kg}}$$

$$= 2336.1 \frac{\text{kJ}}{\text{kg}}$$

3-1

$$q_{3-1} = \Delta U_{3-1}$$

$$= -C_v T_1 (1 - r^{R/C_v + 1})$$

$$= -0.7 \cdot 293 (1 - 5^{0.3/0.7 + 1}) \frac{\text{kJ}}{\text{kg}}$$

$$= 1839 \frac{\text{kJ}}{\text{kg}}$$

1-2

$$W_{1-2} = \Delta U_{1-2}$$

$$= C_v T_1 (r^{R/C_v} - 1)$$

$$= 0.7 \cdot 293 (5^{0.3/0.7} - 1) \frac{\text{kJ}}{\text{kg}}$$

$$= 203.7 \frac{\text{kJ}}{\text{kg}}$$

2-3

$$W_{2-3} = P(a_3 - a_2)$$

$$= R T_1 r^{R/C_v} (r - 1)$$

$$= 0.3 \cdot 293 \cdot 5^{0.3/0.7} (5 - 1) \frac{\text{kJ}}{\text{kg}}$$

$$= 700.8 \frac{\text{kJ}}{\text{kg}}$$

c) $\eta_{th} = 1 - \frac{q_{out}}{q_{in}}$

$$= 1 - \frac{q_{4-1}}{q_{2-3}}$$

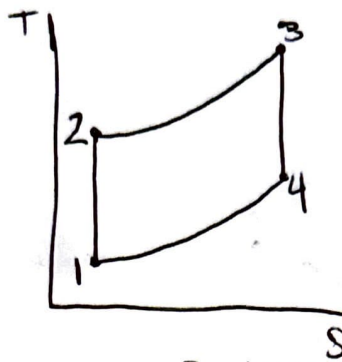
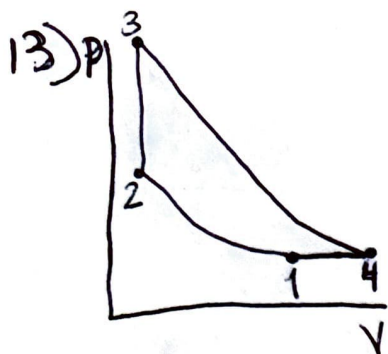
$$= 1 - \frac{0.7(5^{0.3/0.7+1} - 1)}{(0.3 + 0.7)(5 - 1)5^{0.3/0.7}}$$

$$= 0.213$$

Ramzie Abbas

MET 350

HW 1.2



b) from A-17

$$h_1 = 295.17 \frac{\text{kJ}}{\text{kg}}$$

$$Pr_1 = 1.3068$$

$$Pr_2 = \frac{P_2}{P_1} Pr_1$$

$$= \frac{600}{100} \cdot 1.3068$$

$$= 7.8408$$

from A-17 using interpolation:

$$v_2 = 352.29 \frac{\text{kJ}}{\text{kg}}$$

$$T_2 = 490.3 \text{ K}$$

$$v_3 = 1205.41 \frac{\text{kJ}}{\text{kg}}$$

$$Pr_3 = 601.9$$

$$h_4 = 740.92 \frac{\text{kJ}}{\text{kg}}$$

$$P_1 = 100 \quad P_2 = 600 \text{ kPa} \quad P_3 = 1835.6 \text{ kPa} \quad P_4 = 100$$

$$T_2 = 490.3 \text{ K} \quad T_3 = 1500 \text{ K}$$

$$P_3 = P_2 \frac{T_3}{T_2}$$

$$= 600 \cdot \frac{1500}{490.3} \text{ kPa}$$

$$= 1835.6 \text{ kPa}$$

$$Pr_4 = \frac{P_4}{P_3} Pr_3$$

$$= \frac{100}{1835.6} \cdot 601.9$$

$$= 32.79$$

$$w = v_3 - v_2 - (h_4 - h_1)$$

$$= (1205.41 - 352.29 - (740.92 - 295.17))$$

$$= 407.37 \frac{\text{kJ}}{\text{kg}}$$

$$c) \eta_{th} = \frac{w}{q_{in}} = \frac{w}{v_3 - v_2}$$

$$= 407.37$$

$$1205.41 - 352.29$$

$$= 0.478$$

31) $r = 10.5$

$P_1 = 90 \text{ kPa}$

$T_1 = 400^\circ\text{C} = 313 \text{ K}$

$W = 90 \text{ kW}$

$k = 1.4$ (from A-2)

$$\eta = \frac{W}{Q_{in}}$$

$$Q_{in} = \frac{W}{\eta} = \frac{90 \text{ kW}}{0.61}$$

$$= 147.541 \text{ kW}$$

$$= 61\%$$

$$\eta = 1 - \frac{1}{r^{k-1}}$$

$$= 1 - r^{1-k}$$

$$= 1 - 10.5^{1-1.4}$$

$$= \eta = 0.61$$

$$Q_{in} = 147.541 \text{ kW}$$

$$61\%$$