

13.30)

a) mole fraction = ?

$$Y = \frac{P}{P_{\text{tot}}} \quad P_{\text{tot}} = P_{\text{CO}_2} + P_{\text{O}_2} + P_{\text{N}_2} = 100 \text{ kPa}$$

$$Y_{\text{CO}_2} = \frac{P_{\text{CO}_2}}{P_{\text{tot}}} = \frac{20}{100} = 0.20 \quad Y_{\text{O}_2} = \frac{P_{\text{O}_2}}{P_{\text{tot}}} = \frac{40}{100} = 0.40 \quad Y_{\text{N}_2} = \frac{40}{100} = 0.40$$

mass fraction = ?

$$mf = \frac{m}{m_{\text{tot}}}$$

$$m_{\text{CO}_2} = N_{\text{CO}_2} M_{\text{CO}_2} = (20 \text{ kmol}) (44 \text{ kg/kmol}) = 880 \text{ kg}$$

$$m_{\text{O}_2} = N_{\text{O}_2} M_{\text{O}_2} = (40 \text{ kmol}) (32 \text{ kg/kmol}) = 1280 \text{ kg}$$

$$m_{\text{N}_2} = N_{\text{N}_2} M_{\text{N}_2} = (50 \text{ kmol}) (28 \text{ kg/kmol}) = 1400 \text{ kg}$$

$$m_{\text{tot}} = m_{\text{CO}_2} + m_{\text{O}_2} + m_{\text{N}_2} = 3240 \text{ kg}$$

$$mf_{\text{CO}_2} = \frac{m_{\text{CO}_2}}{m_{\text{tot}}} = \frac{880 \text{ kg}}{3240 \text{ kg}} = 0.2716$$

$$mf_{\text{O}_2} = \frac{m_{\text{O}_2}}{m_{\text{tot}}} = \frac{1280 \text{ kg}}{3240 \text{ kg}} = 0.3951$$

$$mf_{\text{N}_2} = \frac{m_{\text{N}_2}}{m_{\text{tot}}} = \frac{1400 \text{ kg}}{3240 \text{ kg}} = 0.4321$$

b) molecular weight = ?

$$M_{\text{A}} = \frac{m_{\text{A}}}{N_{\text{A}}} = \frac{3240 \text{ kg}}{100 \text{ kmol}} = 32.40 \text{ kg/kmol}$$

$$C_v = mf_{\text{CO}_2} C_{v,\text{CO}_2} + mf_{\text{O}_2} C_{v,\text{O}_2} + mf_{\text{N}_2} C_{v,\text{N}_2}$$

(C_v = table A.1)

$$= (0.2716)(0.657) + (0.3951)(0.650) + (0.4321)(0.743)$$

$$C_v = 0.6945 \text{ kJ/kg}\cdot\text{K}$$

$$R = \frac{R_u}{M_{\text{A}}} = \frac{8.314 \text{ kJ/kmol}\cdot\text{K}}{32.40 \text{ kg/kmol}} = 0.2566 \text{ kJ/kg}\cdot\text{K}$$

$$c) C_p = ?$$

$$C_p = C_p + R$$

$$= 16.45 + 25.66$$

$$C_p = 42.1 \text{ kJ/kJ} \cdot K$$

$$K = ?$$

$$K = \frac{C_p}{C_v} = \frac{42.1}{30.45} = 1.370$$

$$13.39) \text{ mass flow rate } (\dot{m}) = ? = \frac{\dot{V}}{v}$$

$$Y_{\text{air}} = 1 - Y_{\text{CH}_4}$$

$$= 1 - 0.15$$

$$Y_{\text{air}} = 0.85$$

$$\dot{m}_m = Y_{\text{CH}_4} \dot{m}_{\text{CH}_4} + Y_{\text{air}} \dot{m}_{\text{air}} \quad \text{Table A-1}$$

$$= (0.15)(16 \text{ kg/kmol}) + (0.85)(28.97 \text{ kg/kmol})$$

$$\dot{m}_m = 27.02 \text{ kg/kmol}$$

$$\dot{V} = \frac{n V_d}{Z} = \frac{(2000 \text{ rpm})(0.005 \text{ m}^3)}{Z} = 7.5 \text{ m}^3/\text{min}$$

$$v = \frac{R \cdot T}{\dot{m}_m} = \frac{(8.314 \text{ kJ/kmol} \cdot \text{K})(293 \text{ K})}{(27.02 \text{ kg/kmol})(60 \text{ s})} = 1.27 \text{ m}^3/\text{kg}$$

$$\dot{m} = \frac{\dot{V}}{v}$$

$$= \frac{7.5}{1.27}$$

$$\dot{m} = 6.66 \text{ kg/min}$$

12.54)

a) $T_n = ?$

Table 9.2

$$M_{Ne} = 20.18 \text{ kg/mol} \quad M_{Ar} = 39.95 \text{ kg/mol}$$

$$N_{Ne} = \frac{P_1 V_1}{R_u T_1} = \frac{(100 \text{ kPa})(1.45 \text{ m}^3)}{(8.314 \text{ J/mol}\cdot\text{K})(293 \text{ K})}$$

$$N_{Ne} = 0.0185 \text{ kmol}$$

$$N_{Ar} = \frac{P_1 V_1}{R_u T_1} = \frac{(200 \text{ kPa})(1.45 \text{ m}^3)}{(8.314 \text{ J/mol}\cdot\text{K})(293 \text{ K})}$$

$$N_{Ar} = 0.0335 \text{ kmol}$$

$$N_{Ne} + N_{Ar} = N_n$$

$$0.0335 + 0.0185 = 0.052 \text{ kmol}$$

$$-Q_{out} = \Delta U = \Delta U_{Ne} + \Delta U_{Ar}$$

$$= [m c_v (T_n - T_1)]_{Ne} + [m c_v (T_n - T_1)]_{Ar}$$

$$-15 \text{ kJ} = (0.0185)(20.18)(0.6179 \text{ kJ/kg}\cdot\text{K})(T_n - 20^\circ\text{C})$$

$$+ (0.0335)(39.95)(0.3122 \text{ kJ/kg}\cdot\text{K})(T_n - 50^\circ\text{C})$$

$$T_n = 16.2^\circ\text{C} + 273 \text{ K}$$

$$T_n = 289.2 \text{ K}$$

b) Final pressure

$$P_n = \frac{N_n R_u T_n}{V_n} = \frac{(0.052 \text{ kmol})(8.314 \text{ J/mol}\cdot\text{K})(289.2 \text{ K})}{1.9 \text{ m}^3}$$

$$P_n = 138.1 \text{ kPa}$$

13.59) $w_{in} = ?$
 $q_{out} = ?$

Table A-1 and 42
 $M_{CH_4} = 16.043 \text{ kg/kmol}$ $M_{C_3H_8} = 44.097 \text{ kg/kmol}$
 $M_{C_4H_{10}} = 58.124 \text{ kg/kmol}$

$$N_{CH_4} = \frac{m_{CH_4}}{M_{CH_4}} = \frac{60 \text{ kg}}{16 \text{ kg/kmol}} = 3.75 \text{ kmol}$$

$$N_{C_3H_8} = \frac{m}{M} = \frac{25 \text{ kg}}{44.1 \text{ kg/kmol}} = 0.5682 \text{ kmol} \quad N_m = 4.5768 \text{ kmol}$$

$$N_{C_4H_{10}} = \frac{m}{M} = \frac{15 \text{ kg}}{58.1 \text{ kg/kmol}} = 0.2586 \text{ kmol}$$

$$N_m = N_{CH_4} + N_{C_3H_8} + N_{C_4H_{10}}$$

$$N_m = 4.5768 \text{ kmol}$$

$$M_m = \frac{m_m}{N_m} = \frac{100 \text{ kg}}{4.5768 \text{ kmol}}$$

$$m_m = 21.85 \text{ kg/kmol}$$

$$R = \frac{R_u}{m_r} = \frac{8.314}{21.85 \text{ kg/kmol}} = 0.3805 \text{ kJ/kg} \cdot \text{K}$$

$$w_{in} = RT \ln \frac{P_2}{P_1} = (0.3805 \text{ kJ/kg} \cdot \text{K})(293 \text{ K}) \ln \left(\frac{100 \text{ kPa}}{1 \text{ atm}} \right)$$

$$w_{in} = 257 \text{ kJ/kg}$$

$$E_{in} = E_{out} = D.E_{system}$$

$$E_{in} - E_{out} = 0$$

$$E_{in} = E_{out}$$

$$m h_1 w_{in} = m h_2 q_{out}$$

$$\begin{aligned} q_{out} + h_2 &= h_1 + w_{in} = h_2 - h_1 - q_{out} \\ &= (c_p (T_2 - T_1)) + q_{out} \\ &= c_p \cdot 0 \cdot q_{out} \end{aligned}$$

$$w_{in} = q_{out}$$

$$257 \frac{\text{kJ}}{\text{kg}} = q_{out}$$