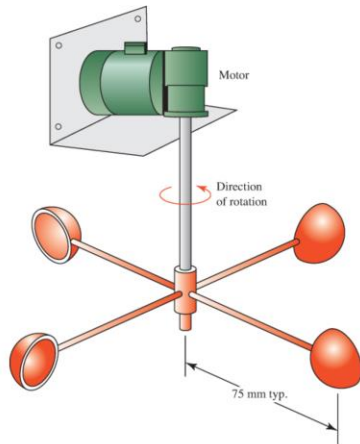


10/21/21

MET 330 HW 2.2 group 11



17.11 A type of level indicator incorporates four hemispherical cups with open fronts mounted as shown in Fig. 17.13. Each cup is 25 mm in diameter. A motor drives the cups at a constant rotational speed. Calculate the torque that the motor must produce to maintain the motion at 20 rpm

$$\omega = \frac{2\pi \text{ rev}}{60 \text{ min}} = \frac{2\pi (20)}{60} \text{ rad/s} = 0.157 \text{ rad/s}$$

$$A = \frac{\pi}{4} (0.025)^2 = 4.91 \times 10^{-4} \text{ m}^2$$

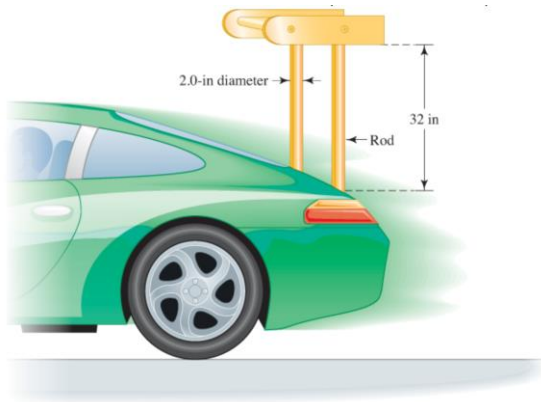
$$N_R = \frac{vD}{v} = \frac{(0.157)(0.025)}{1.6 \times 10^{-5} \text{ m}^2/\text{s}} = 245.31$$

$$F_D = 1.35 \left(\frac{(1.164)(0.157)^2}{2} \right) (4.91 \times 10^{-4}) = 1.06 \times 10^{-5} \text{ N}$$

$v = \text{air at } 30^\circ\text{C}$

$$F_D = 9.51 \times 10^{-6} \text{ N}$$

$$\tau = 4 (9.51 \times 10^{-6}) (0.075) = 2.853 \times 10^{-6} \text{ N}\cdot\text{m}$$



17.14 A wing on a race car is supported by two cylindrical rods, as shown in Fig. 17.15. Compute the drag force exerted on the car due to these rods when the car is traveling through still air at -20°F at a speed of 150 mph.

$$150 \text{ mph} = 67.056 \text{ m/sec}$$

$$a = 32(12) = 64 \text{ in}^2 = 0.0412902 \text{ m}^2$$

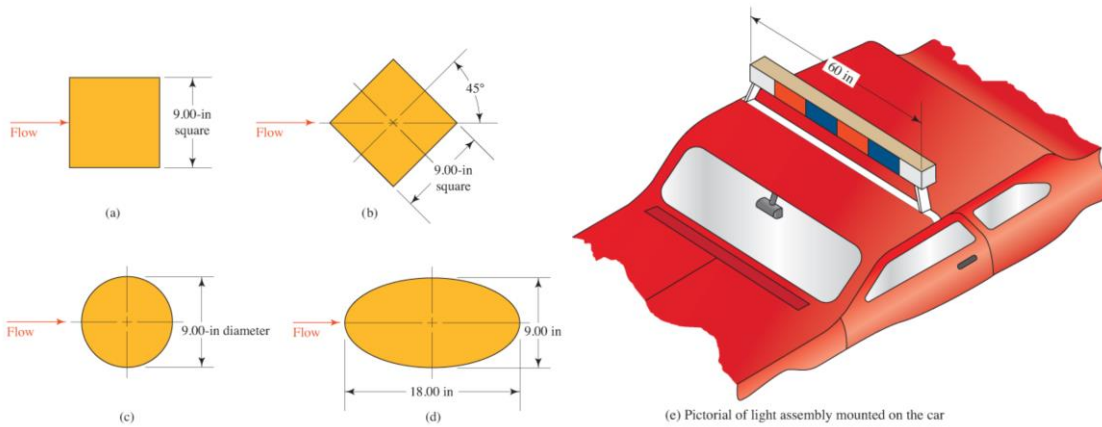


$$F_D = 2 \left(\frac{1}{2} \right) (1.225) (67.056)^2 (0.0412902)$$

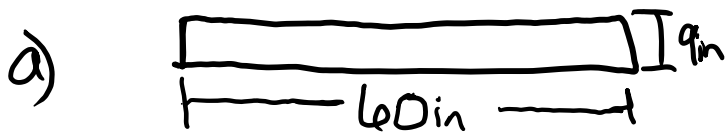
$$F_D = 227.436 \text{ N}$$

$$\rho = 1.225 \text{ kg/m}^3$$

$$F_D = 51.13 \text{ lbf}$$



17.16 The four designs shown in Fig. 17.16 for the cross section of an emergency flasher lighting system for police vehicles are being evaluated. Each has a length of 60 in and a width of 9.00 in. Compare the drag force exerted on each proposed design when the vehicle moves at 100 mph through still air at -20°F .



$$\rho_{\text{air}} = 0.0826$$

$$W = 9 \left(\frac{1}{12} \right) = 0.75 \text{ ft}$$

$$L = 60 \left(\frac{1}{12} \right) = 5 \text{ ft}$$

$$F_D = C_D \left(\frac{\rho V^2}{2} \right) A$$

$$N_R = \frac{VL}{\nu}$$

$$V = 100 (5280) \left(\frac{1}{3600} \right) = 146.67 \text{ ft/s}$$

$$\nu = 1.17 \times 10^{-4} \text{ ft}^2/\text{s}$$

$$\rho = 2.80 \times 10^{-3} \text{ slug/ft}^3$$

$$C_D = 2.10$$

$$N_R = \frac{(146.67)(5)}{1.17 \times 10^{-4}} = 9.401 \times 10^5$$

$$A = 60(9) = 540 \text{ in}^2 \left(\frac{1}{144} \right) = 3.75 \text{ ft}^2$$

$$F_D = 2.10 \left(\frac{(2.8 \times 10^{-3})(146.67)^2}{2} \right) (3.75)$$

$$\underline{\underline{F_D = 237.17 \text{ lb}}}$$

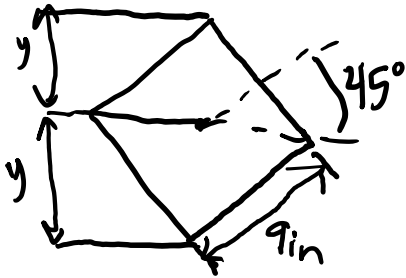
$$b) \sin 45 = y/q$$

$$y = q \sin 45$$

$$y = 6.363 \text{ in} \\ = 0.5302 \text{ ft}$$

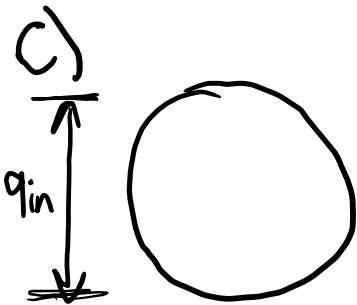
$$A = 5(2)(0.5302) \\ = 5.302 \text{ ft}^2$$

$$C_D = 1.6$$



$$F_D = 1.6 \left(\frac{(2.80 \times 10^{-3})(146.67)^2}{2} \right) (5.302)$$

$$\underline{\underline{F_D = 255.48 \text{ lb}}}$$



$$A = (60)(9) \left(\frac{1}{4} \right) = 3.75 \text{ ft}^2$$

$$N_R = \frac{(146.67)(0.75)}{1.17 \times 10^{-4}} = 9.401 \times 10^5$$

$$C_D = 0.3$$

$$F_D = 0.3 \left(\frac{(2.80 \times 10^{-3})(146.67)^2}{2} \right) (3.75)$$

$$\underline{\underline{F_D = 33.88 \text{ lb}}}$$



$$A = 3.75 \text{ ft}^2$$

$$N_R = \frac{(146.67)(18/12)}{1.17 \times 10^{-4}} = 1.88 \times 10^6$$

$$F_D = 0.25 \left(\frac{(2.80 \times 10^{-3})(146.67)^2}{2} \right) (3.75)$$

$$C_D = 0.25$$

$$\underline{\underline{F_D = 28.23 \text{ lb}}}$$

Table 17.2 Resistance of ships

Ship Type	R_{ts}/Δ
Ocean freighter	0.001
Passenger liner	0.004
Tugboat	0.006
Fast naval ship	0.01–0.12

17.26 A small, fast boat has a specific resistance ratio of 0.06 (see Table 17.2) and displaces 125 long tons. Compute the total ship resistance and the power required to overcome drag when it is moving at 50 ft/s in seawater at 77°F.

$$\Delta = (125 \text{ LT}) \left(\frac{2240 \text{ lb}}{1 \text{ LT}} \right)$$

$$\Delta = 2.8 \times 10^5 \text{ lb}$$

$$R_{ts} = (0.06) (2.8 \times 10^5)$$

$$= \underline{16,800 \text{ lb}}$$

$$P = (16800)(50) = 8.4 \times 10^5 \text{ lb ft/s} \left(\frac{1 \text{ hp}}{550 \text{ lb ft/s}} \right) \quad P = \underline{1527.27 \text{ hp}}$$

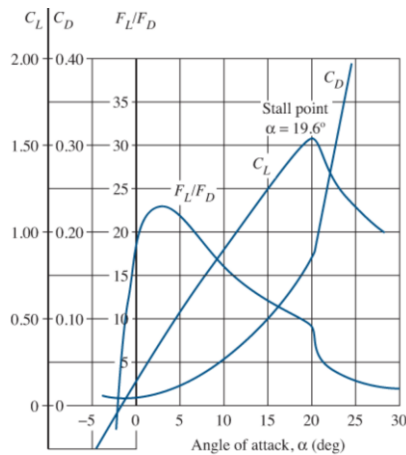


Figure 17.11

Airfoil performance curves.

17.30 For the airfoil with the performance characteristics shown in Fig. 17.11, determine the lift and drag at an angle of attack of 10° . The airfoil has a chord length of 1.4 m and a span of 6.8 m. Perform the calculation at a speed of 200 km/h in the standard atmosphere at (a) 200 m and (b) 10 000 m.

$$V = 200 (1000) (1/3600) = 55.55 \text{ m/s}$$

$$A = (1.4) (6.8) = 9.52 \text{ m}^2$$

$$C_D = 0.05$$

$$C_L = 0.9$$

$$\rho = 1.202 \text{ kg/m}^3$$

A)

$$F_D = 0.05 \left(\frac{(1.202)(55.55)^2}{2} \right) (9.52)$$

$$F_L = 0.9 \left(\frac{(1.202)(55.55)^2}{2} \right) (9.52)$$

$$F_D = 882.77 \text{ N}$$

$$F_D = 0.05 \left(\frac{(0.4135)(55.55)^2}{2} \right) (9.52)$$

$$\underline{F_L = 15889.9 \text{ N}}$$

b)

$$\underline{F_D = 303.683 \text{ N}}$$

$$F_L = 0.9 \left(\frac{(0.4135)(55.55)^2}{2} \right) (9.52)$$

$$\underline{F_L = 5466.29 \text{ N}}$$