

3.6) THE VALUE FOR ABSOLUTE PRESSURE WILL ALWAYS BE GREATER THAN THAT FOR GAGE PRESS. (T/F)

ANS: TRUE; GIVEN THE FORMULA $P_{abs} = P_{gage} + P_{atm}$, P_{abs} WILL ALWAYS BE GREATER.

3.7) AS LONG AS YOU STAY ON THE SURFACE OF THE EARTH, $P_{atm} = 14.7 \text{ psia}$. (T/F)

ANS: FALSE; ATMOSPHERIC PRESSURE CAN CHANGE SIGNIFICANTLY FROM SEA LEVEL TO SOMEWHERE WITH A HIGHER ALTITUDE, e.g. MOUNTAINS

3.8) THE PRESS. IN A TANK IS -55.8 Pa (abs) (T/F)

ANS: FALSE; THE LOWEST POSSIBLE ABSOLUTE PRESSURE IS 0 WHICH WOULD BE A PERFECT VACUUM. GIVEN THIS, ABSOLUTE PRESSURE WILL ALWAYS BE POSITIVE.

3.9) SAME AS ABOVE.

3.10) THE PRESSURE IN A CERTAIN TANK IS -150 kPa (gage) (T/F)

ANS: TRUE; GIVEN THAT IT IS GAGE PRESSURE, IT IS POSSIBLE THAT IT IS A VACUUM.

3.11) IF YOU WERE TO RIDE IN AN OPEN-COCKPIT AIRPLANE TO 9000 ft ABOVE SEA LEVEL, WHAT WOULD THE ATMOSPHERIC PRESSURE BE IF IT CONFORMS TO THE STANDARD ATMOSPHERE?

GIVEN: ALTITUDE OF 9000 ft

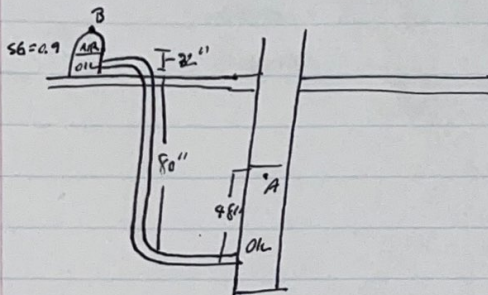
SOLUTION: USING FIGURE E.1 IN BACK OF TEXT, $P_{atm} = \sim 12.7 \text{ psia @ 9000 ft}$

3.13) EXPRESSED IN GAGE PRESSURE, WHAT IS THE PRESSURE AT THE SURFACE OF A GLASS OF MILK?

SOLUTION: ATMOSPHERIC PRESSURE AT SURFACE = $14.7 \text{ psia} \rightarrow 0 \text{ psig} = P_{gage}$

HW 1.2

3.41)



Known: $P_A = 180 \text{ psig}$, $\gamma = \rho \cdot g$, Find P_B

Solution: 1. $\rho_w = 1.94 \frac{\text{slugs}}{\text{ft}^3} \cdot \left(\frac{1 \text{ ft}}{12 \text{ in}} \cdot \frac{1 \text{ ft}}{12 \text{ in}} \cdot \frac{1 \text{ ft}}{12 \text{ in}} \right) = 0.001423 \frac{\text{slugs}}{\text{in}^3}$

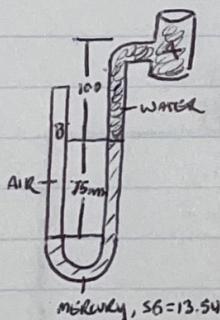
2. $\rho_{oil} = \rho_w \cdot SG_{oil} \rightarrow 0.001423 \frac{\text{slugs}}{\text{in}^3} \cdot (0.9) = 0.001011 \frac{\text{slugs}}{\text{in}^3}$

3. $\gamma_{oil} = \rho_{oil} \cdot g \rightarrow 0.001011 \frac{\text{slugs}}{\text{in}^3} \cdot \left(32.2 \frac{\text{ft}}{\text{s}^2} \cdot \frac{12 \text{ in}}{1 \text{ ft}} \right) = 0.39065$

4. $P_A + \gamma_{oil} \cdot 48 \text{ in} - (\gamma_{oil} \cdot 112 \text{ in}) = P_B$
 $\rightarrow 180 \frac{\text{lb}}{\text{in}^2} + 0.39065(48 \text{ in}) - (0.39065 \cdot 112 \text{ in}) = P_B$

$P_B = 155 \text{ psig}$

3.62)



Known: $P_B = P_{atm} = 0 \text{ Pa (gage)}$

$\gamma_w = 9.81 \frac{\text{KN}}{\text{m}^3}$

$\gamma_m = SG \cdot \gamma_w$

Find $P_A \text{ (kPa)}$

1. $\gamma_m = SG \cdot \gamma_w \rightarrow 13.54(9.81 \frac{\text{KN}}{\text{m}^3}) = 132.827 \frac{\text{KN}}{\text{m}^3}$

2. $P_A = P_B - \gamma_m \cdot (0.075 \text{ m}) - \gamma_w \cdot (0.1 \text{ m})$

$\rightarrow 0 \text{ kPa (gage)} - 132.827 \frac{\text{KN}}{\text{m}^3}(0.075 \text{ m}) - 9.81 \frac{\text{KN}}{\text{m}^3}(0.1 \text{ m})$

$\rightarrow P_A = -10.943 \frac{\text{KN}}{\text{m}^2}$

$P_A = -10.943 \text{ kPa (gage)}$

3.83) Barometer indicates 30.65 in of mercury. Find P_{atm} in psia.

Given: $h = 30.65 \text{ in}$, Known: $P_{atm} = \gamma \cdot h$,

Solution: $\gamma_m = 848.7 \frac{\text{lb}}{\text{ft}^3} \cdot \frac{1 \text{ ft}}{12 \text{ in}} \cdot \frac{1 \text{ ft}}{12 \text{ in}} \cdot \frac{1 \text{ ft}}{12 \text{ in}} = 0.491 \frac{\text{lb}}{\text{in}^3}$

$P_{atm} = \gamma_m \cdot h \rightarrow 0.491 \frac{\text{lb}}{\text{in}^3} \cdot 30.65 \text{ in} = 15.049 \text{ psia}$

$P_{atm} = 15.049 \text{ psia}$

3.90) Pressure in a vacuum chamber is -68.2 kPa , express in mmHg.

Given: $P_v = -68.2 \text{ kPa}$, Find mmHg, Known: $1 \text{ atm} = 760 \text{ mmHg} = P_{\text{atm}}$

Solution:

$$P_{\text{abs}} = P_{\text{atm}} - P_{\text{vacuum}}$$

$$= 760 \text{ mmHg} - P_v \rightarrow P_v = \rho_m \cdot \gamma_w \cdot h$$

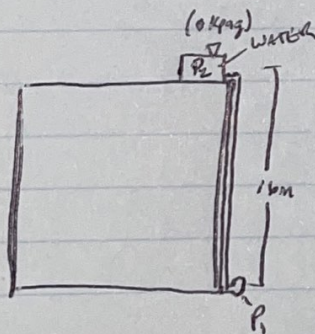
$$\rightarrow h = \frac{P_v}{\rho_m \cdot \gamma_w} = \frac{68.2 \frac{\text{KN}}{\text{m}^2} / 1000}{(13540 \frac{\text{KN}}{\text{m}^3}) \cdot 9.81 \frac{\text{N}}{\text{m}}}$$

$$= 0.513448 \text{ m Hg}$$

$$\rightarrow P_{\text{abs}} = 760 \text{ mmHg} - 513.448 \text{ mmHg}$$

$$= 246.55 \text{ mmHg}$$

3.94)



Given: $h = 16 \text{ m}$, $P_2 = 0 \text{ kPa(g)}$

Known: $\gamma_w = 9.81 \frac{\text{KN}}{\text{m}^3}$

Solution: $P_1 = P_2 + \gamma_w \cdot h$

$$= 0 \text{ kPa(gage)} + 9.81 \frac{\text{KN}}{\text{m}^3} (16 \text{ m})$$

$$P_1 = 156.96 \frac{\text{KN}}{\text{m}^2} \text{ or kPa(gage)}$$