Homework #1.7

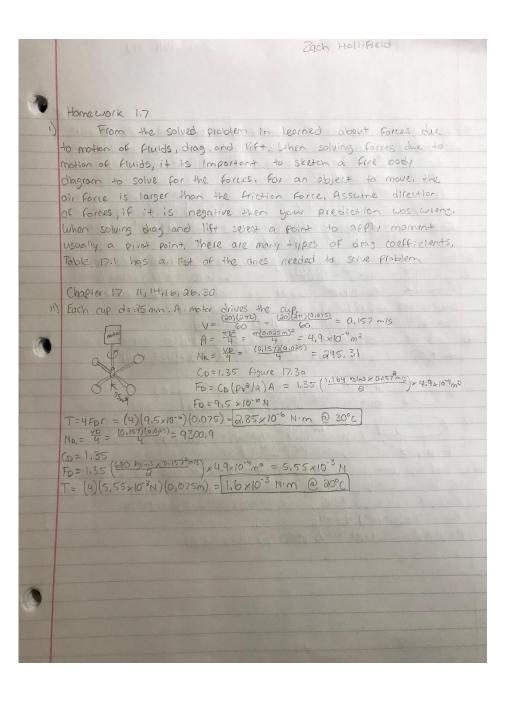
Ch 16 Forces Due to Fluids in motion

Ch 17 Drag and Lift

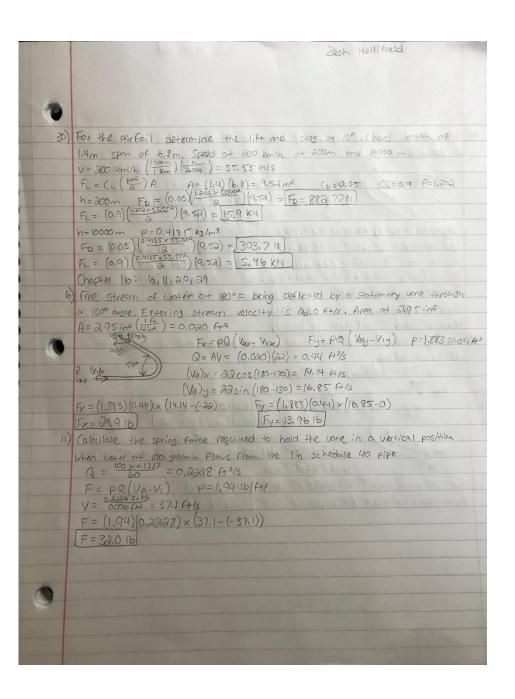
MET 330 Virginia Beach Distance Learning WC2 and Campus

Nathanael Yapnayon - Aaron Jackson - Zach Hollifield

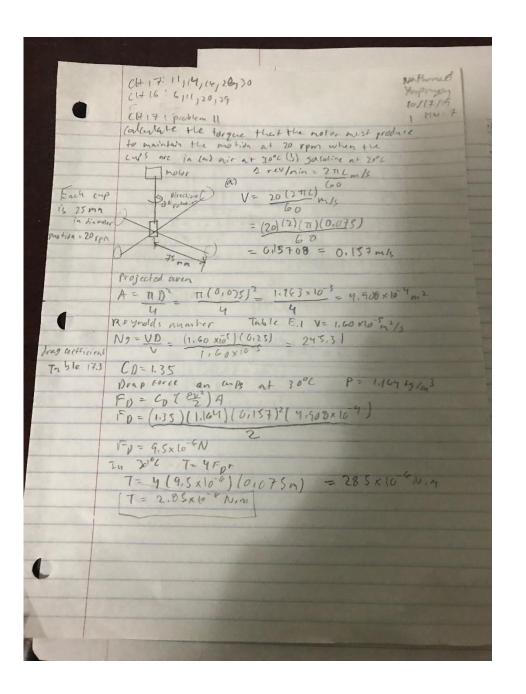
Due Date: 10/17/19

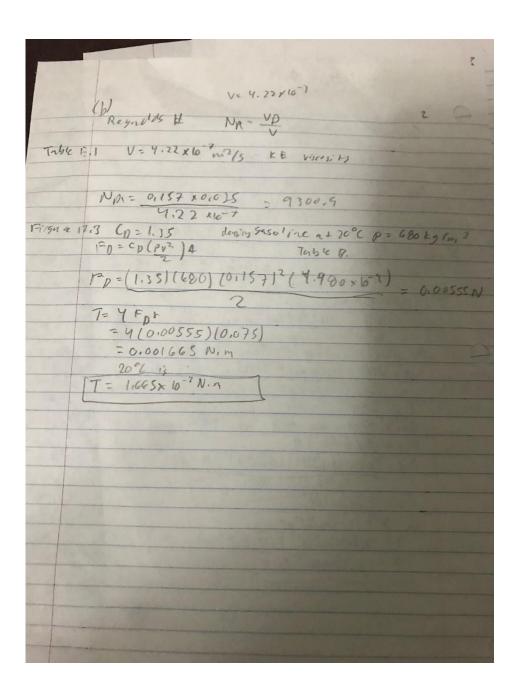


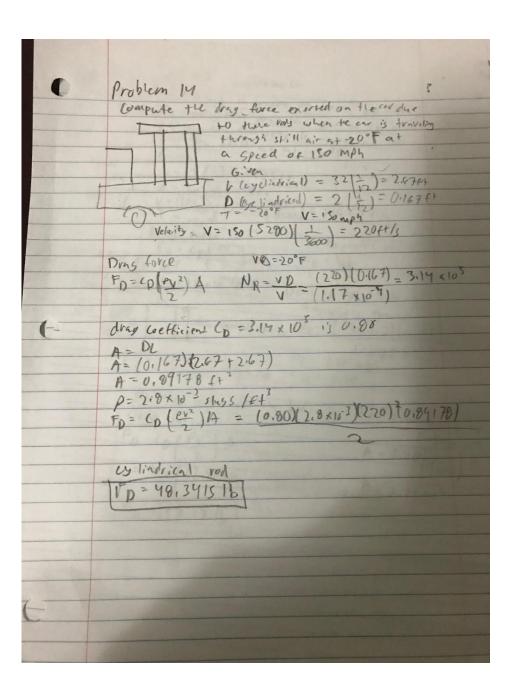
Each Hollifield
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14) The wing on a race car is supported by 2 cylindrical rods, Compute
the drag force exerted on the car to roots at -20° F air of 150 meh
L= 32 in (12in) = 2167 in D=2 in (12in) = 0.167 F+
V=150 mph (5280) (3600) = 220 Ft (5)
V = 150mpn (3000) (3000) 3 0000 1 0000 (1 = 0.80)
$F_{D} = (D(PV^{2}/a)A)$ $N_{R} = \frac{VD}{V}$ $V = 1.17 \times 10^{-4} \frac{G^{2}/5}{V^{2}}$ $N_{R} = \frac{(230 \times 15)(0.187 \text{ ft})}{1_{11} \times 10^{-4} \text{ ft}^{2}/5} = 3.14 \times 10^{-5}$
NR= V=1.17 x 10 1 ft /5 NR= 1,17 x 10 4 ft 2/5 = 5.11 x 10
A=DL = (0.167)(2.67 x2) = 0.89 F+2 P@ 200 F = 2.8 x10 5 m357 F43
A=DL = (0.167)(0.167 x a) = 0.89 f+ 2 P 3 709 F = 2.8 × 10-3 sluss / Fx3 Fb = (0.80) (0.35 1 Fx3 x a) (0.89 f+2) = 48.3 lb
16) Each has a length of 60 in width 9.0 in Compare drag force at 100 mph - 200 pt
16) Fach has a length of 60 in width 9.0 in compare drag force at 100 mph - 20°F a) from NR = $\frac{1}{V} = \frac{(46.67)(0.75)}{1.17\times10^{-47}}$ b = 9x $\frac{1}{4}$ = 0.75 ft $L = 60 \times \frac{1}{12} = 5$ ft
1 NR = 9,4×10 = (0 = 2.1 V=100 × 5280 × 700 = 146.67 6465
0= 200 = 10 = 10 = 10 = 10 = 10 = 10 = 1
$F_0 = (o(R)^2/a) A = 201 \left(\frac{2(4 \times 10^{-3} \times 1446.67^2)}{2}\right)(3.75) = \frac{1}{2}(5)(2 \times 0.5202) = 5.702 \text{ (12)}$
1) T d (2 1-0.5302 C)
b) 3 45° y=95145° = 0.5300 A A = L x dy = (5)(2x6.5300) = 5,300 (40
1
The state of the s
c) Rda $A = (0.75)(5) = 3.75 H2 NR = 9.4 × 105 Cp = 0.3$
Fr = 0.3 (3.75 = 33.9 1b)
D) $A=3.75 Ha$ $NR = \frac{1177 10^{-4}}{2} = 1.88 \times 10^{-6} \text{Co=0.25}$ $R = 0.25 \left(\frac{2.97 10^{-3} \text{MHz} 10^{-3}}{2}\right) 3.75 = 28.3 \text{lb}$
9:0 No = 17210 4 = 1,88×106 (0=0,25
F= 0.35 (27x63x4667) 375 = 28,3 lb
KIRIN TO SURE TO SURE OF THE S
ab) A small fast boot has a specific resistance natto of 0.06 and displaces
135 long tons, Compute the total resistance and power @ 50 ft/s squaler 77° = P=R ₁₅ V R ₁₅ = 0,006 A B = (125) (120,150) = 2,8×10.5 16
P= K+5 V RTS = 9,000 A D= (125) (1 long son) = 0.8×10 10
RES = 0,006 (a,8×105) = 168016 resistance)
P= (1680)(50)= 8,4×104 16-F1/5 (1/16 550 10-16/3) = 152,727 hp



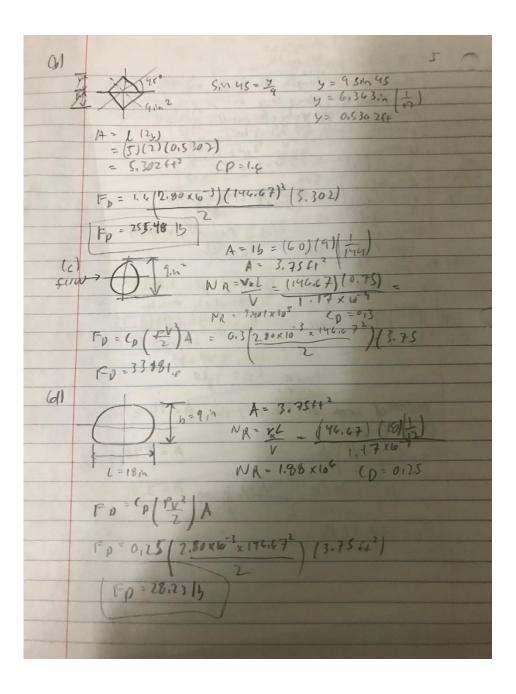
Nathemael Yaprason 10/17/19 What we learned in class 1) When solving for forces we need afree body diagram. This is especially important when solving for static or dynamic problems.
Air force must be larger than its friction cone for the object to move. Drug and lift problems can be solved by apply moment on the point of the drift. Drag letticiants can be found on the tax of in the book. Flow measurement in a pipe is meant to have and out how much money and time. All equations condition Burnelli's equation. Discherge coefficient is dependent on Reynolds Number, Beta is ratio of the 2 diameters.

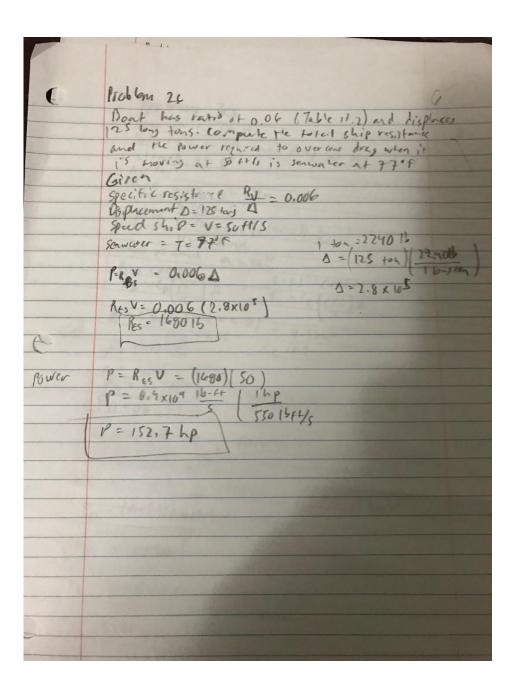




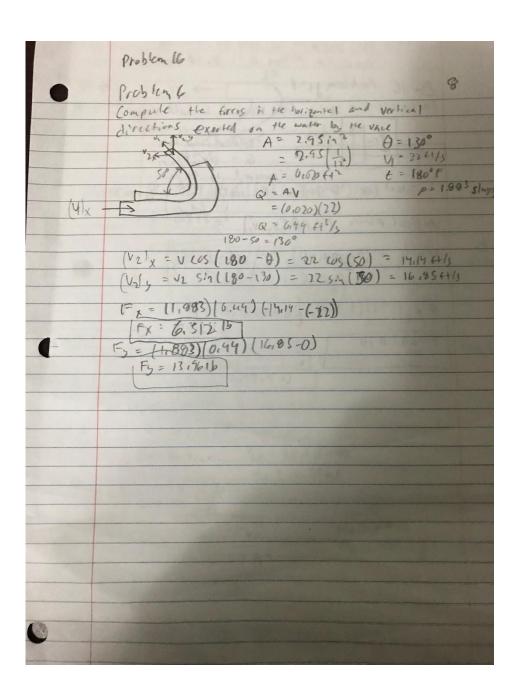


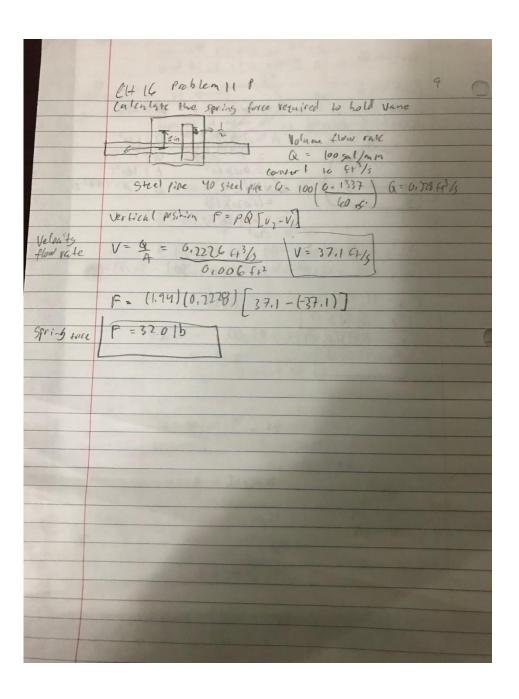
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		Haprayan
- (CH 17 problem lo	4
	Each has a length of Go in and width of good in	
	Compare the drag true exerted on each proposed	
	design when the vehicle moves at 100 mph	
	through still air at -200F	
	Giscon	
squar	b= 9(n)= 0,75++	
1_ength	1= 60/2/25ft	
Velocity	V= (100 mph) (5280) (360) = 146.67 44/5	
	T = -20°F	
	Dras Carle Regners number	
	Fp = Cp A NR = Vel	
Air tubles	6-70°F ()	
A 13(031.)	Vr = 1.17 x10 11/3	
Kinemat. C	Resuld's mumber	
(- (a)	1 9: square No: (146.67)(0.75)	
Slow	Np - 9.201 × 105	
-)00	No= 9,901 x1115	
	37cm = 2,10	
	Cylinder	
	prevented area of the body is A=1b=60	(9)
	Dray firec = se	10,42
	$F_{D} = C_{D} \begin{pmatrix} PV^{2} \\ 2 \end{pmatrix} A \qquad A = 540 \left[\frac{1}{14} \right]$	1=
	= 2.10 (2.80 ×10) (146,47) (3.75) A= 3.75+1	
	A SIR	
	17/1	
	Fo = 237. 17 15	
		4
	123 - 03	
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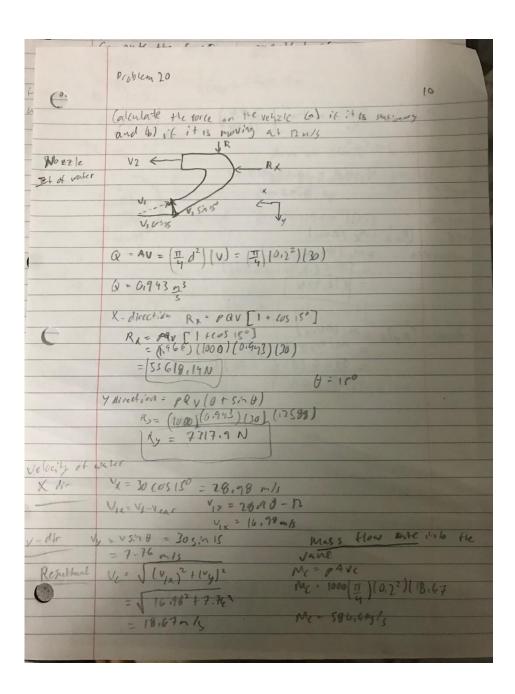




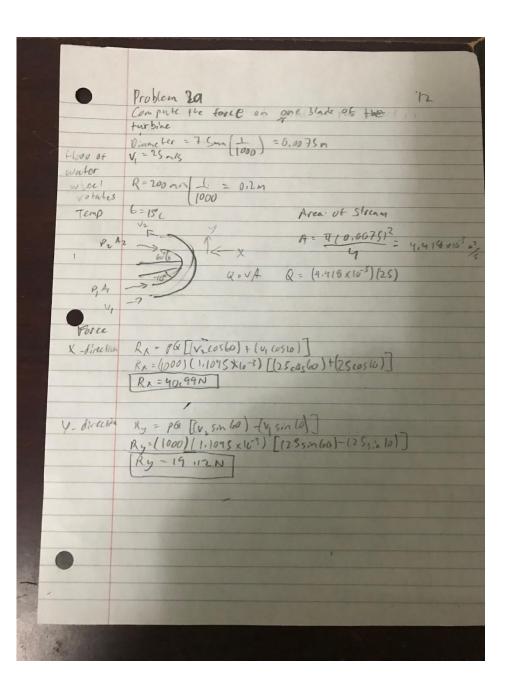
Problem 30 Rese on to 17.11 Determine the lite and drag at an age of attack of 10° Airfoil Chord with (-1.7m 8 pm b = 68 Speed V= (200 km/h) (1000) (300) V= 5555m/s Pray (see the diff $a = 10^{\circ}$ Pray (see the diff $a = 10^{\circ}$ $F_{p} = C_{p}(P^{y^{2}}) A$ $F_{t} = C_{t}(P^{y^{2}}) A$ $A = C_{t} = C_{t}(P^{y^{2}}) A$ $A = C_{t$ angle of attack a = 100 0 $F_{D} = C_{P}(\frac{10^{2}}{2})A = (0.05)(1.202)(55.35)^{2}(9.32)$ $F_{D} = 992.77N$ $F_{L} = C_{L}(\frac{12^{2}}{2})A = (0.09)(1.202)(55.55)^{2}(1.52)$ $F_{L} = 15.999 \times N$ air table dersis (W) when attitude h= 10000m P= 0.4135 ts/3 Fp = (0,05)(0,4135)(55.55)27.52 FD= 303,69 SN Fz - 10.9)(0.9135)(55.55)21952) F = 5744.29N







		4
a	= tom 1/4) = tan (2.46)	
	(41x) (16.59)	
a	= 24.56°	
· · · · · · · · · · · · · · · · · · ·	(par) = Vc cos & = 19.67 cos 9.56	
Vare	= 19.40 m/s	
effective		
Lord 1	(Rx), = My (DVex)	
V 11 -c/2	(Rx)c = Me (DVex) = Me (Vex (pr) - VIX)	
X -arres	- 196.6 1.9 90 - 116.991	
	= 596.6 (19.70 - (16.98)) = 832.97N	
	- 972.17.4	
	(10)	
7- direction	Ry) L=ML (SVEY)	
	= vul veg (par) - Vy	
	= 586.6 (0-(-7.76))	
	(Ry) = me (AVez) = m(Vez(per) - Vy) = 586.6 (0-(-7.76)) = 4552.0N	
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Aaron Jackson

	Hw 1.7 Chp 17: 11, 14, 16, 26, 30 chapter 16: 6, 11, 20, 29 We learned that in order to solve for forces due to motion in fluids that we have to start with a free body diagram. Drog is the force on a body coused by the fluid that resists motion in the direction of the body. Lift is a force coused by the fluid in a direction perpendicular to the direction of travel of the body. Solving drag and 11+ the distant drag coefficients as there are multiple.
	chapter 17
17.11	D= 25 mm
0	1= 20.27.0.075 60 = 0.157 m/s
	A= TO2 = T. 0.025 m2 = 4.9.10-4 m2
	Co=1/35
	FO= (0 Pr2/2)+
	= 1.35 · 1.164 × 2/m3 · 0.1572 m15 , 49×10-4,2
6	
	0=9,5 x10-6N
T	= 4Par
1	= 4.9,5×10-4.0.075=2.45.10-6 N.m & 30°C
1.08	= 10 = (0.157) (0.025) - 9300,9
	4

CD = 1.35 FO = 1.35 (600 149/m2.0.1572 4/3), 49 410-4 m2 = 5.55 x10-3 T= 4 .5.55 . 10-3 N D.675m = 1, 6x 10-3 Nin 2200 17.14 D=Zin=0.1678+ L= 32in= 2.67in V= 150 man = 220 7+19 Fo= Co (Py 2/2) A NR = ND N=1117 . 10 -4 24 2/5 5.7015 - COLO = 10 = A = 0,89 FO = 0.60 . 2.8.10-3 = 1030/ft 2. 2203 . 0.49 == = = 48.3 16 17.16 à) NR= VL = 144.67.0.75 6=9.1 = 0.75+1 V 1.17+10-4 /= 100mph = 146.67+2/5 NR=9.4 ×105 CD=2.1 P=2.40×10-3 =100×1/6+2 A=LB=0,75,5=3,75 st2 FO=LO(Pr2/2)A=21 2.8×10 3,146.672, 3.75=237.216 6) y=9 sin45° = 0.5302 ft A= 1.24 = 5. 6.5302 = 5.300x CO=1.6 (2.8.10-3 × 146.47) 15.302 = 255.4616

CA-0.75.5 = 3.75 Ft2 FD= 0.3 , 2.8×10-3.146.67, 3.75 = 33.916 O A=3,75 F+2 NR - 146.67 . 18/12 - 1.88.104 CO=025 1.17.10-4 Fo = 0.25: 2.8.10-3.144.47 .3,75 = 28,3 16 17.76 P=R.3V RTS = 0,006 A B= 125, 2240 16 - 7.8 × 105 16 Ris = 0.006. 2,4,105 = 1680 16 resistence P=1660.50=6.4×10+16. F+15 1he = 152.727 hp 17.30 V= 200 xm/h = 55.55 mls A= 1,4.6,8=9.52 m2 FCELL DV2 . A CD =0.05 A=1,4,6,4=9,52m2 h= 200m Fo=0.05 1,202,56.552, 915L -Fo=682,77 FL = 0.9 . 1.202 . 55.552 . 9.5L = 15.9KN

h= 10000 m p= 0.4175 45/m? FO= 0.05 . 0.4135 . 55.55 . 9.56 = 303.7 N FL= 0.9 , 0.4135 , 55.55 , 9,52 = 5.46 KN Chaples 16 16.6 fx = pol/2x-Vix) Q= AV = 0.020 122 = 0.44 871/5 1/245 225 IN (160-BO) = 16165 FT/5 Fy = 1, 683 . 0.44 . 16, 65 . 0 will Q= 100.0.1337 F= pa(v2-V1) p=10+16+17 V=02226 ft 31 0.0006 = 37.1 ==1.94.0.226 .37.1- 37.1 = (32/6)

= T. (0.2) . 30 = 0.143 m /1 JTG.942 + 7.762 = 16,67 MS Vx = VLOSO = 30 COSIS = 28.98 m/3 Vy= Vsin 0 = 30 sin 15 = 7. Trents Ryle = 556.6 10-7.76)=(4552.0N) 0-75mm=0,0075m (6,29 Rx=PD (V200560° +V, COS 100) Ry= Palveziolos + V, siarge) 11,525 A/S /2:25 M/S BY = 1000. 1.1045, 10-3 25 500 60 + 25 costo (Rx=41N) BY = 1000. 1.1045, 10-3 25 500 + 255, n 10 (Ry=14.1N)