

Colby Watts
Professor Ayala
01/30/24
MET 350 HW 1.3

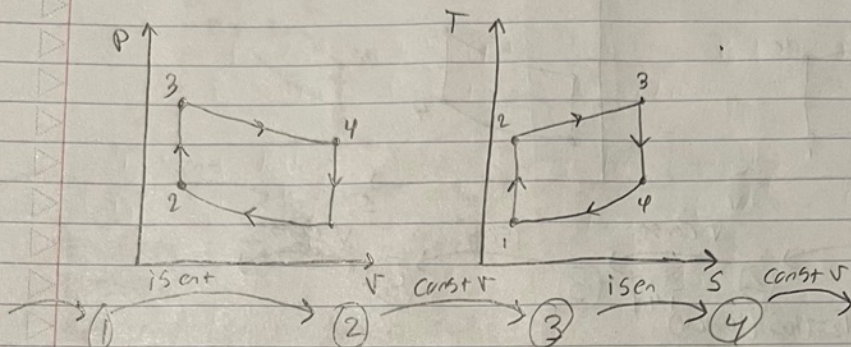
HW 1.3

9.33

$$\frac{P_2}{P_1} = 8 = \frac{P_4}{P_3}$$

$$R = 0.287 \text{ kJ/kg}\cdot\text{K}$$

$$Q_{in} = 750 \text{ kJ/kg}$$



$$P_1 = 95 \text{ kPa}$$

$$P_2 = 1705 \text{ kPa}$$

$$V_3 = 77.6 \text{ v}$$

$$T_4 = 774.5 \text{ K}$$

$$T_1 = 300 \text{ K}$$

$$V_2 = 77.60$$

$$T_3 = 1539 \text{ K}$$

$$P_3 = 3898.37 \text{ v}$$

$$P_2 = 8 \cdot \frac{673.1 \text{ K}}{300}$$

$$V_{r2} = \frac{V_2}{V_1} = \frac{V_3}{V_4}$$

$$V_{r1} = \frac{V_1}{V_2} = \frac{V_4}{V_3}$$

$$u_1 = 214.07$$

$$T_2 = 673.1 \text{ K}$$

$$P_3 = 3898.4 \text{ kPa}$$

$$u_2 = 491.2 \text{ kJ/kg}$$

$$Q_{in\ 2-3} = u_3 - u_2 = Q_{in\ 2-3} + u_2 = 750 + 491.2 = 1241 \text{ kJ/kg}$$

$$T_3 = 1539 \text{ K}$$

$$P_3 = \frac{T_3}{T_2} P_2$$

$$\text{Table } u_4 = 571.69 \text{ kJ/kg}$$

$$V_{r4} = \frac{V_1}{V_2} \cdot V_{r3} = 8 \cdot 6.581 = 52.70$$

$$T_4 = 774.5 \text{ K}$$

$$W_{net} = Q_{in} - Q_{out} = 750 - 357.62 = 392.4 \text{ kJ/kg}$$

$$\eta_{th} = \frac{W_{net}}{Q_{in}} = \frac{392.4}{750} \cdot 100 = 52.32\%$$

$$V_{max} = R \frac{T_1}{P_1} = 0.287 \cdot \frac{300}{95} = 0.09063 \text{ m}^3/\text{kg}$$

$$V_{min} = V_2 = 77.60$$

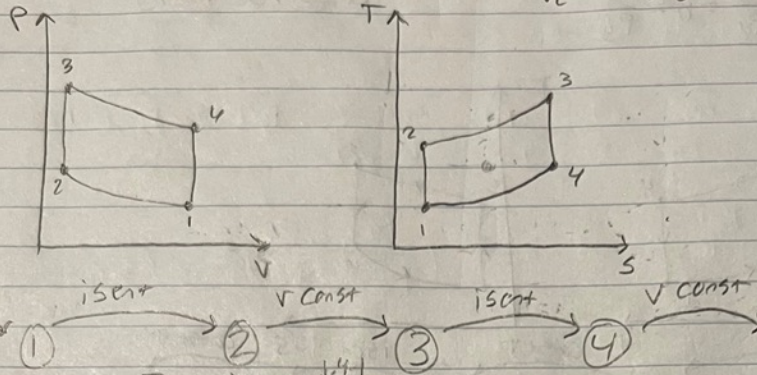
$$\text{mean eff } P = \frac{W_{net}}{(V_1)(1 - \frac{1}{r})} = \frac{392.4}{(0.09063)(1 - \frac{1}{8})} = 495 \text{ kPa}$$

E. J. W. H.

9.36 -

Engine speed = 2500 rpm

$V_c = 9.8\%$



$P_1 = 14 \text{ psia}$ $T_2 = (105)(10.204)^{1.411}$

$T_1 = 105^\circ\text{F}$ $T_2 = 265.89^\circ\text{F}$ $T_3 = 2400^\circ\text{F}$

$r_c = \frac{V_1}{V_2} = \frac{100}{10.2} = 9.8$

$Q_{in} = m_{air} (T_3 - T_2) = 0.176 (2400 - 265.88) = 366.214 \frac{\text{Btu}}{\text{lbm}}$

$\eta_{th} = \left(1 - \frac{1}{r^{(\gamma-1)}}\right) = 1 - \frac{1}{(10.204)^{0.411}} = 60.5\%$

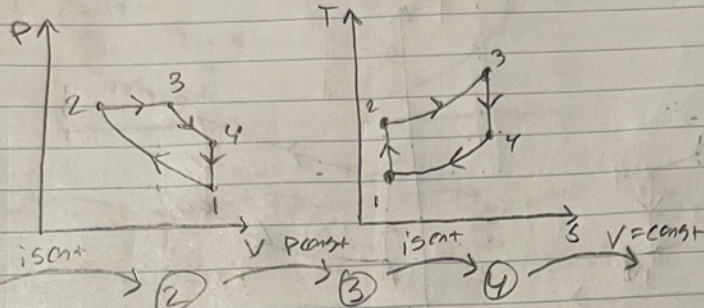
$\eta = \frac{N}{2} = \frac{2500}{2} = 1250$

$0.605 = \frac{W_{net}}{366.214}$ $W_{net} = 221.5 \text{ Btu/lbm}$

Power = $\frac{W_{net} \cdot n \cdot \dot{m}}{60} = \frac{(221.5 \text{ Btu/lbm})(1250 \text{ rpm})(6)}{60}$

$P = 27,687 \text{ ft}^2/\text{s}$

9.46

 $\Gamma = 16$ $\Gamma_c = 2$ 

$$P_1 = 95 \text{ kPa}$$

$$T_2 = 300 \text{ K}$$

$$T_3 = T_2$$

$$T_1 = 300 \text{ K}$$

(a) Temp after heat addition

$$T_3 = T_2 \left(\frac{\Gamma^k - 1}{\Gamma^k - 1} \right) = 300 \left(\frac{16^{1.4} - 1}{2^{1.4} - 1} \right) = 1725 \text{ K}$$

$$(b) \eta_{th} = 1 - \frac{1}{\Gamma^k} \left(\frac{\Gamma^k - 1}{\Gamma - 1} \right) = 1 - \frac{1}{16^{1.4}} \left(\frac{16^{1.4} - 1}{2^{1.4} - 1} \right) \cdot 100 = 56.3\%$$

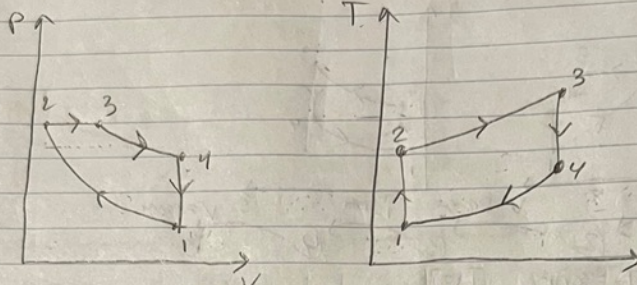
$$(c) \text{ mean eff } P = \frac{P_1 V_1}{\Gamma - 1} \cdot \frac{\Gamma^k - 1}{k \cdot \Gamma^k - 1} = \frac{(95)(1000)}{16 - 1} \cdot \frac{2^{1.4} - 1}{(1.4)(16^{1.4} - 1)} = 675.9 \text{ kPa}$$

9.57

$$R = 0.287$$

$$\gamma = 2.2$$

$$\gamma_c = 1.8$$



$P_1 = 92 \text{ kPa}$
 $T_1 = 343 \text{ K}$
 $V_1 = \frac{RT_1}{P_1} = \frac{(0.287)(343)}{92} = 1.044$
 $T_2 = 343 \cdot 2.2^{1.8-1} = 1181 \text{ K}$
 $V_2 = \frac{1.044}{2.2} = 0.463$
 $P_3 = 7347.64 \text{ kPa}$
 $T_3 = 2125.9 \text{ K}$
 $V_3 = 0.083034$
 $T_4 = \left(\frac{V_3}{V_1}\right)^{1.4-1} = 780.95 \text{ K}$
 $T_4 = 780.95 \text{ K}$

$$W_{1-2} = -C_v(T_2 - T_1) = -0.718(1181 - 343) = -601.68 \text{ kJ/kg}$$

$$P_3 = \frac{RT_3}{V_3} = \frac{(0.287)(2125.9)}{0.083034} = 7347.64 \text{ kPa}$$

$$V_3 = \gamma_c(V_2) = (1.8)(0.04613) = 0.083034$$

$$T_3 = \frac{P_3 V_3}{R} = \frac{(7347.64)(0.083034)}{0.287} = 2125.9 \text{ K}$$

$$W_{2-3} = P_3(V_3 - V_2) = 7347.64 \text{ kPa}(0.083034 - 0.04613) = 271.1 \text{ kJ/kg}$$

$$W_{net} = m(W_{1-2} + W_{2-3} + W_{3-4} + W_{4-1})$$

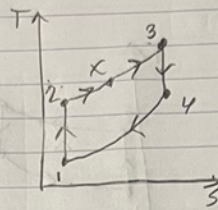
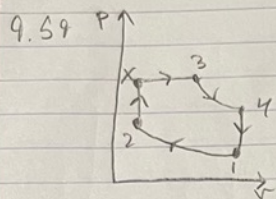
$$W_{net} = (2.36)(-601.68 + 271.1 + 965.6 + 0)$$

$$W_{net} = 1.501 \text{ kJ}$$

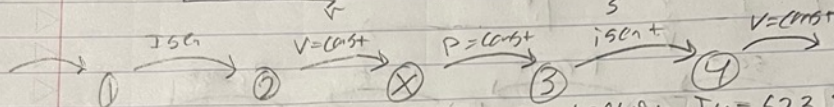
$$W_{3-4} = C_v(T_3 - T_4) = 0.718(2125.9 - 780.95)$$

$$W_{3-4} = 965 \text{ kJ/kg}$$

$$W_{4-1} = 0$$



$$\gamma = 1.5 \quad \gamma_c = 1.4$$



$$P_1 = 14.2 \text{ PSia} \quad T_2 = 677.39 \text{ K} \quad T_x = 965.13 \text{ K} \quad T_3 = 1,351.19 \text{ K} \quad T_4 = 523.27 \text{ K}$$

$$T_1 = 297 \text{ K}$$

$$T_2 = T_1 (r)^{\gamma_c}$$

$$T_x = (1.1)(T_2)$$

$$T_3 = P/T_x$$

$$T_4 = (1.1)(1.4)^{\gamma_c}(297)$$

$$\text{Specific heat addition} = C_v(T_x - T_2) + C_p(T_3 - T_x)$$

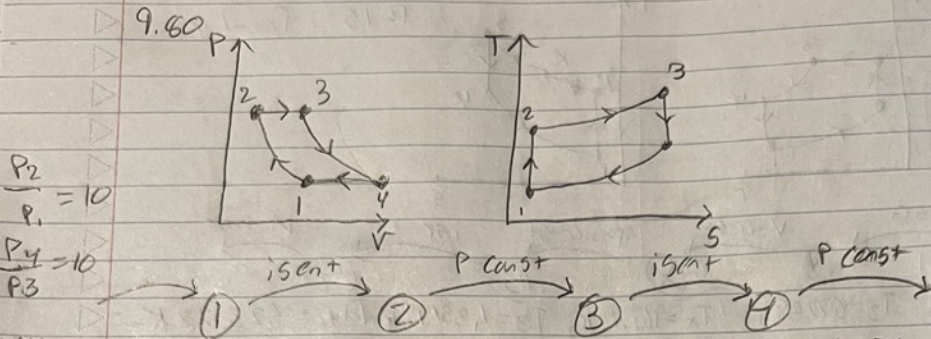
$$(0.718)(965.13 - 677.39) + (1.005)(1,351.19 - 965.13)$$

$$Q_{in} = 450.986 \text{ kJ/kg}$$

$$W_{net} = Q_{in} - Q_{out} = 450.986 - C_v(T_4 - T_1)$$

$$450.986 - 0.718(523.27 - 297) = 288.42 \text{ kJ/kg}$$

$$\eta_{th} = \frac{W_{net}}{Q_{in}} = \frac{288.42}{450.986} \cdot 100 = 63.98\%$$



Used Compressor $T_1 = 520 R = 290 K$ $P_2 = 12.311$ $T_2 = 2000 R = 1100 K$ $P_{r4} = 16.71$

A-17 $h_1 = 290.16 \text{ kJ/kg}$ $h_3 = 1161.07 \text{ kJ/kg}$

$P_{r1} = 1.2311$

$P_{r2} = \frac{P_2}{P_1} P_{r1} \checkmark$

$P_{r3} = 167.1$

$P_{r4} = \frac{P_4}{P_3} (167.1) = 16.71$

Compressor $\rightarrow w_c = h_2 - h_1 = (560.62 - 290.16) = 270.46 \text{ kJ/kg}$

Turbine $\rightarrow w_t = h_3 - h_4 = (1161.07 - 611.431) = 549.62 \text{ kJ/kg}$

$\rightarrow q_{in} = h_3 - h_2 = (1161.07 - 560.62) = 600.45 \text{ kJ/kg}$

$B.W.R. = \frac{w_t}{w_c} = \frac{270.46 \text{ kJ/kg}}{549.62 \text{ kJ/kg}} \cdot 100 = \boxed{49.21\%}$

$\eta_{th} = \frac{w_t - w_c}{q_{in}} = \boxed{46.49\%}$