

Chapter 15.4

PROBLEM: A SHARP-EDGED ORIFICE IS PLACED IN A 10-IN.-DIAMETER PIPE CARRYING AMMONIA. IF THE VOLUME FLOW RATE IS $25 \frac{\text{GAL}}{\text{MIN}}$, CALCULATE THE DEFLECTION OF A WATER MANOMETER

(A) IF ORIFICE DIAMETER IS 1.0 IN

(B) IF THE ORIFICE DIAMETER IS 7.0 IN

THE AMMONIA HAS A SPECIFIC GRAVITY OF 0.83 AND A DYNAMIC VISCOSITY OF $2.5 \times 10^{-6} \frac{\text{lb}}{\text{ft}^2 \cdot \text{s}}$

$$d_2 = \frac{1.0}{12} = 0.083 \text{ ft} \quad A_2 = \frac{\pi d_2^2}{4} = \frac{\pi (0.083 \text{ ft})^2}{4} = 0.00541$$

$$(A) \quad d_1 = \frac{10}{12} = 0.83 \text{ ft} \quad A_1 = \frac{\pi d_1^2}{4} = \frac{\pi (0.83 \text{ ft})^2}{4}$$

$$\gamma_{\text{AMMONIA}} = 0.83$$

$$\gamma_w = (62.4 \frac{\text{lb}}{\text{ft}^3}) (1.0) \quad A_1 = .541 \text{ ft}^2$$

$$B = \frac{d}{D} = \frac{1.0}{10} = 0.1$$

$$\mu_{\text{AMMONIA}} = (0.83) (62.4 \frac{\text{lb}}{\text{ft}^3}) = 52.416 \frac{\text{lb}}{\text{ft}^3}$$

$$Q = \left(\frac{25 \text{ gal}}{\text{min}} \right) \left(\frac{1 \text{ ft}^3}{7.48 \text{ gal}} \right) = .0557 \frac{\text{ft}^3}{\text{s}}$$

$$V_1 = \frac{Q}{A_1} = \frac{.0557 \frac{\text{ft}^3}{\text{s}}}{.541 \text{ ft}^2} = .103 \frac{\text{ft}}{\text{s}}$$

$$N_R = \frac{V_1 D}{\nu} = \frac{(.103 \frac{\text{ft}}{\text{s}}) (0.83)}{1.5 \times 10^{-6} \frac{\text{ft}^2}{\text{s}}} = 57491.3$$

$$V_1 .103 = (.592) \sqrt{\frac{2(32.2) h \left(\left(\frac{62.4}{52.416} \right) - 1 \right)}{\left(\frac{.541 \text{ ft}^2}{.00541 \text{ ft}^2} \right)^2 - 1}} = \sqrt{\frac{72.3 h}{9999}} = (.103) / (.592)$$

$$h = \frac{(.174)^2}{.00123} = 24.6 \text{ ft}$$

$$V_1 = C \sqrt{\frac{2g(P_1 - P_2)}{\gamma}}$$

$$\left(\frac{A_1}{A_2} \right)^2 - 1$$

$$= C \sqrt{\frac{2gh(\gamma_m/\gamma) - 1}{\left(\frac{A_1}{A_2} \right)^2 - 1}}$$

$$V = \frac{Q}{P} = \frac{(25 \times 10^{-6} \frac{\text{lb}}{\text{ft}^2 \cdot \text{s}})}{(52.416 \frac{\text{lb}}{\text{ft}^3})}$$

$$32.2$$

$$V = 1.5 \times 10^{-6} \frac{\text{lb}}{\text{s}}$$

$$(B) d_2 = \frac{7.0}{12} = .583 \text{ ft}$$

$$A_2 = \frac{\pi}{4} (.583 \text{ ft})^2 = .267 \text{ ft}^2$$

$$A_1 = .541 \text{ ft}^2$$

$$B = \frac{7.0}{10.0} = .7$$

$$\gamma_{\text{AMMONIA}} = 52.416 \frac{\text{lb}}{\text{ft}^3}$$

$$\gamma_w = 62.4 \frac{\text{lb}}{\text{ft}^3}$$

$$Q = .0557 \frac{\text{ft}^3}{\text{s}}$$

$$V_1 = .103 \frac{\text{ft}}{\text{s}}$$

$$NR = 57491.3$$

$$C = .592$$

$$.103 = (.592) \sqrt{\frac{2(32.2)h \left(\frac{62.4}{52.416} \right) - 1}{\left(\frac{.541 \text{ ft}^2}{.267 \text{ ft}^2} \right)^2 - 1}}$$

$$\sqrt{\frac{12.3h}{3.1}} = (.103)(.592)$$

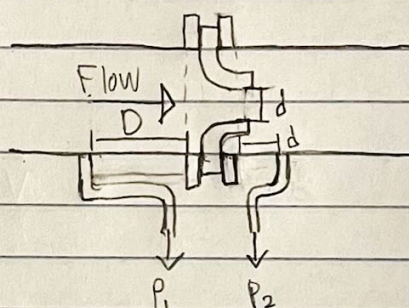
$$h = (.174)^2 / (3.97) = .0076 \text{ ft}$$

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Problem:

A flow nozzle is to be installed in a 5 in Type K copper tube carrying linseed oil at 77°F. A mercury manometer is used to measure the pressure difference across the nozzle when expected range of flow rate is 700 gpm to 1000 gpm. The manometer scale ranges from 0 to 8.0 in of mercury. Determine the appropriate diameter of the nozzle.

5 in Type K copper
 $D = 0.4004 \text{ ft}$
 $A_1 = 0.1259 \text{ ft}^2$

 $\Rightarrow$ 

700 gpm - 1000 gpm
 449 gpm

$$V_1 = C \sqrt{\frac{2gh(\gamma_m/\gamma_f - 1)}{(A_1/A_2)^2 - 1}}$$

0 - 8 in = 0 - 0.667 ft $Q_{1-2} = 1.56 \text{ ft}^3/\text{s} - 2.23 \text{ ft}^3/\text{s}$

$$A_2 = \sqrt{\frac{2gh(\gamma_m/\gamma_f - 1)}{\left(\frac{Q}{A_1 C}\right)^2 + 1}}$$

$A_1 = 0.1259 \text{ ft}^2$

$V_1 = \frac{Q}{A_1}$

$\gamma_m = 844.9 \text{ lb/ft}^3$

$\gamma_f = 58.0 \text{ lb/ft}^3$

$h = 0.667 \text{ ft}$

$C = 0.92$

$$\frac{\pi d^2}{4} = 0.0768 \text{ ft}^2$$

$d = 0.313 \text{ ft}$

$C_{new} = 0.91$

Within 1%

$$N_R = \frac{V D}{\nu} = \frac{(2.23)(0.4004 \text{ ft})}{3.84 \times 10^{-4}}$$

$N_R = 3486.1$

$d = 0.313 \text{ ft or } 3.76 \text{ in}$

15.15

HW 2.3

A pitot-static tube is inserted into a duct carrying air at standard atmospheric pressure & a temperature of 50°C . A differential manometer reads 0.24 in of water. Calculate the velocity of the flow

$$\text{temp: } 50^{\circ}\text{C} = 122^{\circ}\text{F}$$

$$g = 32.2 \text{ ft/s}^2$$

$$\gamma_{\text{air at } 122^{\circ}\text{F}} \approx 0.06846 \text{ lb/ft}^3$$

$$h_{\text{H}_2\text{O}}: 0.24 \text{ in} = 0.02 \text{ ft}$$

$$\gamma_{\text{H}_2\text{O}} = 62.4 \text{ lb/ft}^3$$

$$V_1 = \sqrt{2gh(\gamma_{\text{H}_2\text{O}} - \gamma) / \gamma}$$

$$V_1 = \sqrt{2(32.2 \text{ ft/s}^2)(0.02 \text{ ft})(62.4 \text{ lb/ft}^3 - 0.06846 \text{ lb/ft}^3) / 0.06846 \text{ lb/ft}^3}$$

$$V_1 = \sqrt{64.4 \text{ ft/s}^2 (0.02 \text{ ft})(62.33154 \text{ lb/ft}^3) / 0.06846 \text{ lb/ft}^3}$$

$$V_1 = \sqrt{96.34 / 0.06846 \text{ lb/ft}^3}$$

$$V_1 = \sqrt{1407.25}$$

$$V_1 = 37.51 \text{ ft/s}$$