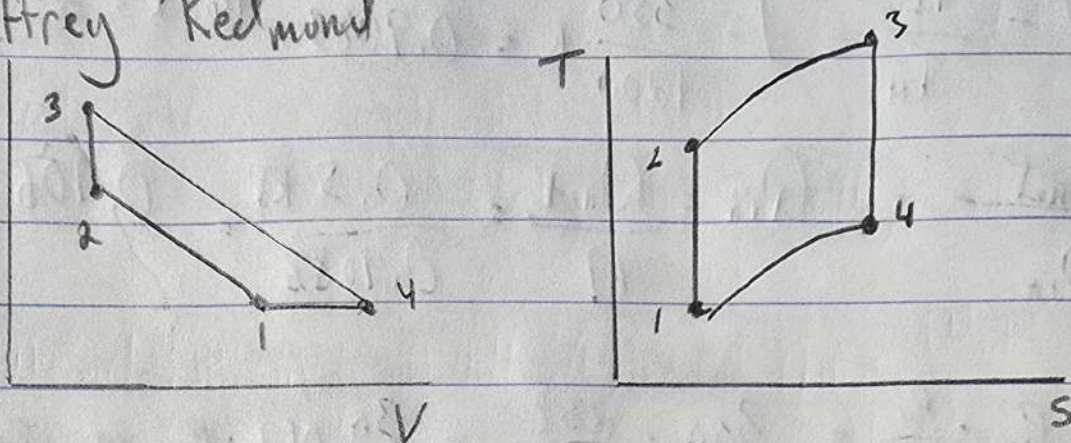


HW 1.2

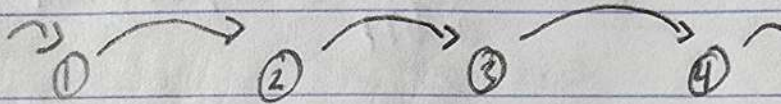
Jeffrey Redmond

9-13

(a)



(b)



$$P_1 = 100 \text{ kPa} \quad P_2 = 600 \text{ kPa} \quad P_3 = 1102 \text{ kPa} \quad P_4 = 100 \text{ kPa}$$

$$T_1 = 501 \text{ K} \quad T_2 = 817 \text{ K} \quad T_3 = 1500 \text{ K} \quad T_4 = 861 \text{ K}$$

$$\frac{T_2}{T_1} = \frac{P_2}{P_1} \frac{K-1}{K}$$

$$P_3 = P_2 \frac{T_3}{T_2}$$

$$T_2 = T_1 \cdot \frac{P_2}{P_1} \frac{K-1}{K}$$

$$P_3 = 600 \cdot \frac{1500}{817} \text{ kPa} = 1102 \text{ kPa}$$

$$T_2 = 501 \text{ K} \cdot \frac{600}{100} \frac{1.371-1}{1.371}$$

$$T_4 = T_3 \cdot \frac{P_4}{P_3} \frac{K-1}{K}$$

$$T_2 = 817 \text{ K}$$

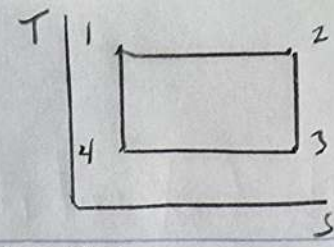
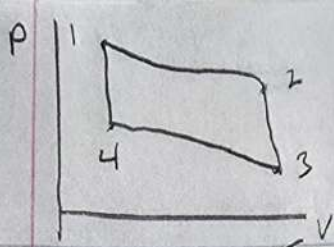
$$T_4 = 1500 \left(\frac{100}{1102} \right)^{\frac{1.371-1}{1.371}} = 861 \text{ K}$$

$$Q_{in} = c_{v,avg} (T_3 - T_2) = 0.8865 (1500 - 817) = 605 \text{ kJ/kg}$$

$$Q_{out} = c_{v,avg} (T_4 - T_1) = 1.058 (861 - 501) = 381 \text{ kJ/kg}$$

$$W_n = Q_{in} - Q_{out} = 605 - 381 = 224 \text{ kJ/kg}$$

$$(c) \eta_{th} = \frac{W_n}{Q_{in}} = \frac{224}{605} = 0.37 = 37\%$$



$350 \text{ K} = 2.379 \text{ MPa}$
 $1200 \text{ K} = 238 \text{ KPa}$

9-18 $\eta = 1 - \frac{T_L}{T_H} = 1 - \frac{350}{1200} = 0.7083$

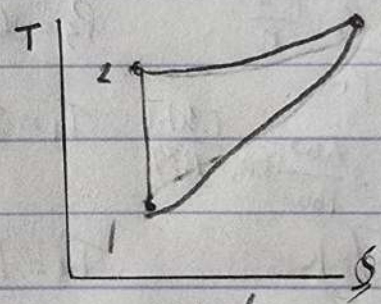
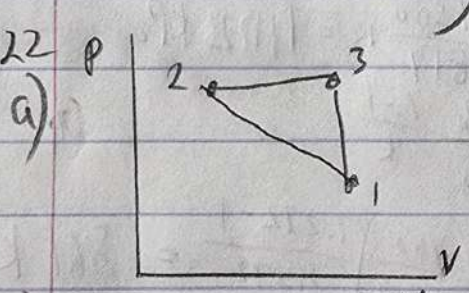
b) $\eta = \frac{W_{net}}{Q_{in}} = \frac{Q_{out}}{Q_{in}} = \frac{0.5 \text{ kJ}}{0.7083} = 0.706 \text{ kJ}$

a) $P_1 = P_4 \cdot \frac{P_1}{P_4} = 300 \cdot \frac{238}{2.379} = 30,012 \text{ KPa} \text{ or } 30 \text{ MPa}$

c) $m = \frac{W}{R \ln\left(\frac{P_4}{P_3}\right) (T_H - T_L)} = \frac{0.5}{0.287 \ln\left(\frac{300}{150}\right) (1200 - 350)}$

$m = 0.00296 \text{ kg}$

9-22



b) $c_p - c_v = R$ $c_p = 0.3 + 0.7 = 1 \text{ kJ/kg}\cdot\text{K}$ $\frac{c_p}{c_v} = \frac{1}{0.7} = 1.428$

$\text{1} \rightarrow \text{2}$
 $\frac{T_2}{T_1} = \left(\frac{V_2}{V_1}\right)^{\gamma}$ $T_2 = 293(5)^{1.428-1} = 583.47 \text{ K}$
 $W_{1-2} = c_v(T_1 - T_2) = 0.7(293 - 583.47) = -203.335 \frac{\text{kJ}}{\text{kg}}$

$\text{2} \rightarrow \text{3}$
 $\frac{T_3}{T_2} = \frac{V_3}{V_2}$ $T_3 = 583.47 \cdot 5 = 2917.4 \text{ K}$

$W_{2-3} = c_p(T_3 - T_2) - c_v(T_3 - T_2)$
 $= 2333.93 - 0.7(2917.4 - 583.47) = 700.179 \frac{\text{kJ}}{\text{kg}}$

$q_{3-1} = c_v(T_3 - T_1) = 0.7(2917.4 - 293) = 1837.687 \frac{\text{kJ}}{\text{kg}}$

9-23: $q_{2-3} = c_p(T_3 - T_2)$
 $= 1(2917.4 - 583.47) = 2333.93 \frac{\text{kJ}}{\text{kg}}$
 $W_{3-1} = 0$

c) $\eta = 1 - \frac{q_{out}}{q_{in}} = 1 - \frac{1837.687}{2333.93} = 21.28\%$

$$Q-3) \quad \eta = 1 - \frac{1}{r^{k-1}} = 1 - \frac{1}{10.5^{1.4-1}} = 60.95\%$$
$$Q = \frac{W}{\eta} = \frac{90}{0.6095} = 147.64 \text{ kW}$$

$$\eta = 1 - \frac{1}{8.5^{1.4-1}} = 57.51\%$$

$$Q = \frac{90}{0.5751} = 156.48 \text{ kW}$$